



777 Aviation Dr
Camarillo, Ca 93010
USA

March 5th, 2026
Revision F

Maintenance Support Document

ETM1000 - AS350 - ICA

Instructions for Continued Airworthiness;

Installation of an AKV, Inc Exceedence and Trend Monitoring System Kit P/N ETM1000 in the AS350 with the Arriel 1 or LTS101 engines

STC Number: SR02413LA

S/N _____

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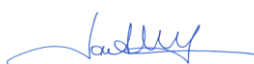
REVISIONS INTRODUCTION

The latest revision of this document is indicated by the highest revision letter as listed below in the Revision History and List of Effective Pages. Changes to the current revision will be indicated within the document by change bars. (Reference Section 1.) The entire document will be reprinted to reflect the current revision level.

REVISION HISTORY

Instructions for Continued Airworthiness; AKV Inc Exceedence and Trend Monitoring System ETM1000			ETM1000-ICA
Rev.	Date	Revision Description	Approval
NC	January 11 th , 2012	Initial Release	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED
A	April 5 th , 2013	Added Appendix A Data	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED
B	June 6 th , 2013	Changed Caution and Warning LED Cap Color to White and Blue	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED
C	October 10 th , 2014	Added -2 configuration without pushbutton / indicators and audio side-tone	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED

Continued

Instructions for Continued Airworthiness; AKV Inc Exceedence and Trend Monitoring System ETM1000			ETM1000-ICA
Rev.	Date	Revision Description	Approval
D	September 22 nd , 2018	- Added cycle counting option with remote display or 3 rd party Flightcell DZMx. -Updated App. A data. Changed to refer to the User Manual - Adjusted Appendix list due to App. A change - Change ICA doc from ETM1000-ICA to ETM1000-AS350-ICA and added engine model applicability - Updated App. D Drawings to include change for remote display and Flightcell DZMx interface - Various changes throughout	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED 
E	November 5 th , 2020	- Added new AA battery holder in place of coin cell and updated Appendix D drawings	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED 
F	March 5 th , 2026	- Updated Appendix D drawings to MDL Rev R	PREPARED J. Gunn
			CHECKED J. Gunn
			APPROVED 

LIST OF EFFECTIVE PAGES

All pages are revised when any page is changed so that all pages maintain the same revision level.

PAGE	REVISION	DATE
All	NC	January 11th, 2012
All	A	April 5 th , 2013
9	B	June 6 th , 2013
2, 3, 4, 9, 10, 11, 23, 27	C	October 10 th , 2014
ALL	D	September 22 nd , 2018
10 and Appendix D	E	November 5 th , 2020
Appendix D	F	March 5 th , 2026

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1. INTRODUCTION

This AKV, Inc maintenance support document provides instructions for the continued airworthiness of the AKV Exceedence and Trend Monitoring System ETM1000. The basis for this document is 14 CFR 27.1529 and Appendix A to Part 27.

1.1 Service Difficulties

Technical Assistance can be provided by:

AKV, Inc.
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Camarillo, CA 93010
Tel 805-437-1739
Fax 805 437-1783
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Web: www.akvinc.com

1.2 Warnings, Cautions, and Notes

WARNING

**FAILURE TO FOLLOW INSTRUCTIONS GIVEN IN A WARNING MAY
RESULT IN PERSONAL INJURY OR DEATH.**

CAUTION

*FAILURE TO FOLLOW INSTRUCTIONS GIVEN IN A CAUTION MAY
RESULT IN DAMAGE TO THE HELICOPTER*

Note

A note includes supplemental data about the procedure, practice, condition, etc., for the maintenance task you are about to perform.

1.3 Language

This manual is written to the Simplified English (SE) specification. This International Aerospace Maintenance Language Specification is important to maintenance personnel whose first language is not English.

1.4 References

The use of parentheses throughout this document is for denoting or depicting a reference to other sections, items, details, etc., the intent of which is to further identify or clarify existing information.

1.5 Revision to ICA

Revisions to this document and the documents listed within it are distributed to the operators who have this modification installed on the subject aircraft, either in electronic or paper format. **Contact AKV for available revision changes.**

1.6 Abbreviations and Definitions

A&P	Airframe & Powerplant Mechanic
CFR	Code of Federal Regulations
RFM	Rotorcraft Flight Manual
RFMS	Rotorcraft Flight Manual Supplement
FAR	Federal Aviation Regulations

2. AIRWORTHINESS LIMITATIONS

This section is FAA approved and specifies the inspections, and other maintenance, which are required under 14 CFR 43.16 and 91.403, unless an alternative program has been FAA approved. There are “No airworthiness limitations associated with this type design change”.

3. DESCRIPTION

3.1 ETM1000 Configurations

The ETM1000 is available in two configurations, a -1 or -2 configuration. In either configuration the ETM1000 enclosure is mounted on the pilot side below the instrument panel. It is electrically spliced to the existing signal generators for Tq, MGT, N1, N2 and Nr behind the instrument panel and at the rear of each indicator. The system receives airspeed indication in the form of a calibrated 40kt airspeed switch connected via a “tee” fitting in the pitot line and mounted in the forward section of the instrument panel area. It provides a signal to the ETM1000 that the A/C is operating 40Kt or greater and is utilized for Tq exceedence criteria. An OAT probe is mounted on the belly and the ETM1000 LRU has a pressure altitude (PA) sensor built-in. The OAT and PA sensor are used for the Delta Ng and power check recording.

As a -1 configuration, two (2) instrument panel mounted pushbutton / indicators (annunciators) and audio side-tone are provided for overall status and alert indication to the pilot. A flip guard is installed on the P-PWR/CHK (left side switch) to prevent inadvertent operation. The “P” in the P-PWR/CHK and P-MUTE white nomenclature indicates the pushbutton feature.

NOTE: Cycle counting is available as an option on the -1 configuration via the AKV remote display or displayed on the 3rd party Flightcell DZMx. The **LRU must be Rev P** or greater to allow for using the cycle AKV remote display and **software v72.0 or greater** must be installed for the cycle counting option.

As a -2 configuration, there are no pushbutton / indicators or audio side-tone supplied.

NOTE: The cycle counting / remote display or Flightcell DZMx interface options are not available with the -2 configuration.

The system date and time is backed up with an externally mounted battery for easy access which is **replaced annually**. A Single power source via a 2 amp circuit breaker supplies power to the ETM1000.

ETM1000
ENCLOSURE
(LRU)
LOCATION

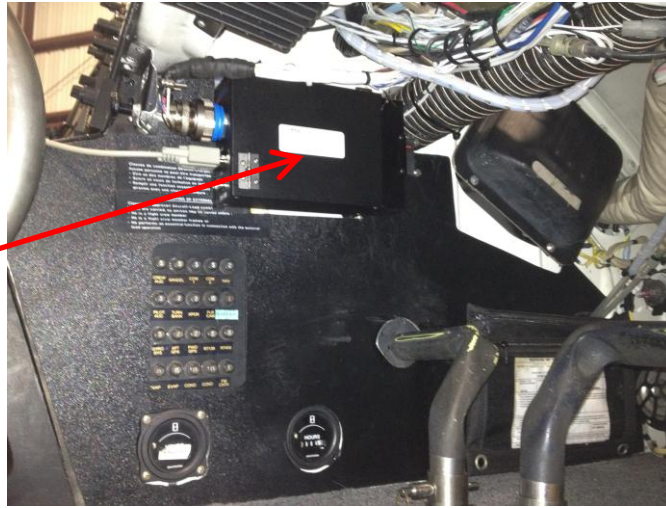


Figure 1 - Installed view of the ETM1000 LRU

-1 CONFIGURATION
ONLY. TYPICAL
PUSHBUTTON /
INDICATOR
LOCATION



Figure 2 - Installed view of the ETM1000 pushbutton / annunciator

(OPTIONAL) CYCLE
COUNTING
REMOTE DISPLAY
LOCATION



Figure 3 - Installed view of the (optional) cycle counting remote display



Figure 4 –(Optional) Flightcell DZMx with cycle counting display

3.2 ETM1000 Operation

The ETM1000 is designed to continuously monitor and record all engine and drive train parameters at 1Hz (1/sec). For values in excess of the airframe and engine operating limitations during an exceedance, it is recorded at 5hz (5/sec). The 1Hz data referred to as a Run Log (RL) is used during post flight for graphing normal flight operations. Exceedence data is also graphed.

As a -1 configuration system health and exceedence status indication via white caution and blue warning indicators is provided to the pilot via two (2) pushbutton / annunciators mounted on the instrument panel. An audible side tone for the pilot's headset is also provided for caution and warning indication. Audible muting and a power check recording feature is provided as part of the pushbuttons. In the -1 configuration **only**, the optional cycle count is available.

NOTE: When using the optional cycle counting feature with either the AKV remote display or Flightcell DZMx then refer to the **ETM1000 Arriel 1 or LTS101 Cycle Counting Operating Specification** document.

As a -2 configuration there is no pilot interface and the ETM1000 system is essentially a black box recorder.

All data is written to a removable 2GB SD Card. Exceedances are backed up in the ETM1000 flash memory in case the SD card is missing.

NOTE: For additional information, refer to the **ETM1000 User Manual**.

4. MAINTENANCE INSTRUCTIONS

4.1 Routine Cleaning & Maintenance

Routine Cleaning & Maintenance of the ETM1000 Routine maintenance may be required as the result of an inspection. Re-tighten loose fasteners to the standard torque values shown in Table 1 and Table 2. Perform routine cleaning to remove contaminants from the ETM1000. Remove grease, fungus, and ground-in dirt from equipment and mounting brackets using a clean, soft cloth dampened with mild soap and warm water; avoid damaging the coating.

4.2 Tools

The following tools will be necessary to maintain the AKV ETM1000 Installation:

- A Laptop PC utilizing Windows XP or later operating system
- RS-232 cable as supplied with the ETM1000 kit
- USB adaptor as supplied with the ETM1000 kit
- Pitot test set (user supplied)

4.3 Overhaul

There are no component overhaul requirements for this type design change. Under normal operating conditions, the ETM1000 will not require component overhaul. All parts and assemblies are designed to be replaced "On Condition". Any maintenance needed to the ETM1000 or its installation beyond that described in this document requires that the components be removed and returned to AKV, Inc. The decision to return damaged components may be a subjective one and should be made by a qualified A & P Mechanic.

4.4 Component Retirement/Retirement Life

The ETM1000 has been designed with components that have a virtually unlimited life span. However, it is anticipated that some components may require replacement at some time during the service life of the helicopter. The decision to replace these parts is a subjective one and should be made by the operator or an A & P Mechanic.

Bolts -- Steel Tension		Bolts -- Steel Tension		Bolts -- Aluminum								
AN 3 – AN 20		MS 20004 – MS 20024		AN 3DD – AN 20DD								
AN 42 – AN 49		NAS 144 – NAS 158		AN 173DD – AN 186DD								
AN 73 – AN 81		NAS 583 – NAS 590		AN 509DD								
AN 173 – AN 186		NAS 144 – NAS 158		AN 525D								
MS 20033 – MS 20046		NAS 144624 – NAS 644		MS 27039D								
MS 20073		NAS 1303 – NAS 1320		MS 24694DD								
MS 20074		NAS 172		-----								
AN 509 NK9		NAS 174		-----								
MS 24694		NAS 517		-----								
AN 525 NK525		-----		-----								
MS 27030		-----		-----								
		Steel Shear Bolt										
		NAS 464										
Nuts		Nuts		Nuts								
Steel Tension	Steel Shear	Steel Tension	Steel Shear	Aluminum Tension	Aluminum Shear							
AN 310	AN 320	AN 310	AN 320	AN 365D	AN320D							
AN 315	AN 364	AN 315	AN 364	AN 310D	AN 364D							
AN 363	NAS 1022	AN 363	NAS 1022	NAS 1021D	NAS 1022D							
AN 365	MS 17826	AN 365	MS 17826	-----	-----							
NAS 1021	MA 20364	MS 17825	MS 20364	-----	-----							
MS 17825	-----	MS 20365	-----	-----	-----							
MS 21045	-----	MS 21045	-----	-----	-----							
MS 20365	-----	NAS 1021	-----	-----	-----							
MS 20500	-----	NAS 679	-----	-----	-----							
NAS 679	-----	NAS 1291	-----	-----	-----							
FINE THREAD SERIES ¹												
Nut-Bolt Size	Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.	
	Min.	Max	Min.	Max	Min.	Max	Min.	Max	Min.	Max	Min.	Max
8 - 36	12	15	7	9	-----	-----	-----	-----	5	10	3	6
10 - 32	20	25	12	15	25	30	15	20	10	15	5	10
1/4 - 28	50	70	30	40	80	100	50	60	30	45	15	30
5/16 - 24	100	140	60	85	120	145	70	90	40	65	25	40
3/8 - 24	160	190	95	110	200	250	120	150	75	110	45	70
7/16 - 20	450	550	270	300	520	630	300	400	180	280	110	170
1/2 - 20	480	690	290	410	770	950	450	550	280	410	160	260
9/16 - 18	800	1000	480	600	1100	1300	650	800	380	580	230	360
5/8 - 18	1100	1300	660	780	1250	1550	750	950	550	670	270	420
3/4 - 16	2300	2500	1300	1500	2650	3200	1600	1900	950	1250	5560	880
7/8 - 14	2500	3000	1500	1800	3550	4350	2100	2600	1250	1900	750	1200
1 - 14	3700	4500	2200	3300	4500	5500	2700	3300	1600	2400	950	1500
1 1/8 - 12	5000	7000	3000	4200	6000	7300	3600	4400	2100	3200	1250	2000
1 1/4 - 12	9000	11000	5400	6600	11000	13400	6600	8000	3900	5600	2300	3650

Table 1: Recommended torque values for nut-bolt combinations: Fine Thread Series

¹ Torque values without lubrication

Bolts -- Steel Tension		Bolts -- Steel Tension		Bolts -- Aluminum								
AN 3 – AN 20		MS 20004 – MS 20024		AN 3DD – AN 20DD								
AN 42 – AN 49		NAS 144 – NAS 158		AN 173DD – AN 186DD								
AN 73 – AN 81		NAS 583 – NAS 590		AN 509DD								
AN 173 – AN 186		NAS 144 – NAS 158		AN 525D								
MS 20033 – MS 20046		NAS 144624 – NAS 644		MS 27039D								
MS 20073		NAS 1303 – NAS 1320		MS 24694DD								
MS 20074		NAS 172		-----								
AN 509 NK9		NAS 174		-----								
MS 24694		NAS 517		-----								
AN 525 NK525		-----		-----								
MS 27030		-----		-----								
		Steel Shear Bolt										
		NAS 464										
Nuts		Nuts		Nuts								
Steel Tension	Steel Shear	Steel Tension	Steel Shear	Aluminum Tension	Aluminum Shear							
AN 310	AN 320	AN 310	AN 320	AN 365D	AN320D							
AN 315	AN 364	AN 315	AN 364	AN 310D	AN 364D							
AN 363	NAS 1022	AN 363	NAS 1022	NAS 1021D	NAS 1022D							
AN 365	MS 17826	AN 365	MS 17826	-----	-----							
NAS 1021	MA 20364	MS 17825	MS 20364	-----	-----							
MS 17825	-----	MS 20365	-----	-----	-----							
MS 21045	-----	MS 21045	-----	-----	-----							
MS 20365	-----	NAS 1021	-----	-----	-----							
MS 20500	-----	NAS 679	-----	-----	-----							
NAS 679	-----	NAS 1291	-----	-----	-----							
COARSE THREAD SERIES ²												
Nut-Bolt Size	Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.		Torque Limits In-lbs.	
	Min.	Max	Min.	Max	Min.	Max	Min.	Max	Min.	Max	Min.	Max
8 - 32	12	15	7	9	-----	-----	-----	-----	-----	-----	-----	-----
10 - 24	20	25	12	15	-----	-----	-----	-----	-----	-----	-----	-----
1/4 - 20	40	50	25	30	-----	-----	-----	-----	-----	-----	-----	-----
5/16 - 18	80	90	48	55	-----	-----	-----	-----	-----	-----	-----	-----
3/8 - 16	160	185	95	110	-----	-----	-----	-----	-----	-----	-----	-----
7/16 - 14	235	255	140	155	-----	-----	-----	-----	-----	-----	-----	-----
1/2 - 13	400	480	240	290	-----	-----	-----	-----	-----	-----	-----	-----
9/16 - 12	500	700	300	420	-----	-----	-----	-----	-----	-----	-----	-----
5/8 - 11	700	900	420	540	-----	-----	-----	-----	-----	-----	-----	-----
3/4 - 10	1150	1600	700	950	-----	-----	-----	-----	-----	-----	-----	-----
7/8 - 9	2200	3000	1300	1800	-----	-----	-----	-----	-----	-----	-----	-----
1 - 8	3700	5000	2200	3000	-----	-----	-----	-----	-----	-----	-----	-----
1 1/8 - 8	5500	6500	3300	4000	-----	-----	-----	-----	-----	-----	-----	-----
1 1/4 - 8	6500	8000	4000	5000	-----	-----	-----	-----	-----	-----	-----	-----

Table 2: Recommended torque values for nut-bolt combinations: Coarse Thread Series

² Torque values without lubrication

5. REQUIRED INSPECTIONS

5.1 Cycle Counting Daily Coherence Check (Installed as an option)

The following procedure applies to either use of the AKV remote display or 3rd party Flightcell DZMx display and provides an approved method for conforming with the SAFRAN “coherence check” as specified in the SAFRAN maintenance manual or with reference to SAFRAN / Turbomeca General Service Letter No. 2283/04 5th issue. It will provide a means to determine that the cycle counting function is recording cycles within your normal mission range.

Procedure: To understand your normal mission range, determine what the total manual count was for the last 10 flights. Add all 10 flights together and then divide by 10. The result is the average consumed cycles to which your normal range is determined by adding +/- 10%. Perform this function separately for Ng and Np.

After the last flight of the day, compare the recorded cycles from the cycle counter display with your normal mission range values. This method will insure that there is no significant discrepancy which would be evident with either a digital display of zero “flt” cycles or “flt” cycles that are in excess of what is the normal mission range. In the event that either of these situations is observed, then perform the Annual Inspection of the cycle counter as described below.

5.2 Annual Inspection

This section contains the time limit intervals and requirements for the scheduled and conditional inspections for the ETM1000 Installation. The inspection items are determined through experience, tests, and the judgment of mechanics and engineers. Every calendar and daily inspection is a visual and thorough inspection to determine the airworthiness of the ETM1000. Qualified persons must perform the inspections in accordance with standard aircraft practices and the applicable maintenance manuals.

An Annual inspection using this **ETM1000-ICA** document (Instructions for Continued Airworthiness), is required. When using **ETM1000-ICA** to perform the required inspections, use the “Inspection Task Description” forms below to record the appropriate data.

Any reference data needed to perform this inspection, other than this document, will be listed in the “Data Reference” column of the inspection forms.

The inspection task to be completed is described in the “Inspection Task Description” column of the inspection form.

When each task is satisfactorily completed, the mechanic performing the inspection task signs the “Mechanic” column for that task.

If the inspection task fails the inspection, the reason for the failure must be determined and remedial action taken before the “Mechanic” block can be signed off. Remedial action may involve cleaning, verifying correct operation, adjusting the tightness of fasteners and hardware, sending the assembly to the manufacturer for overhaul or replacement, etc. Once the appropriate remedial action has been taken and the inspection task has been satisfactorily performed, the mechanic may sign off the “Mechanic” block.

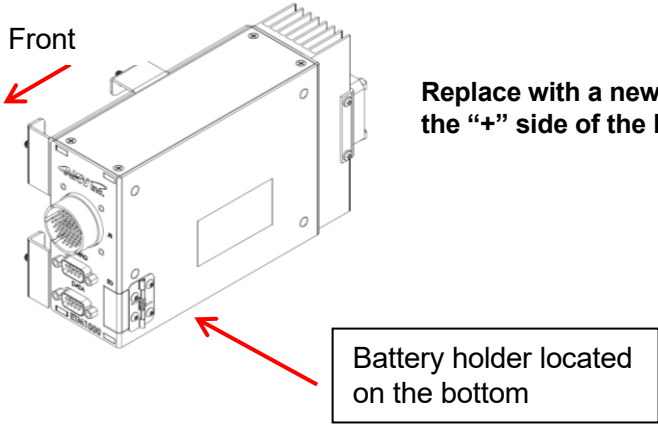
Once all tasks are signed off in the “Mechanic” Column, the inspection form may be signed off at the top of the form and the inspection has then been completed.


5.3 Special Inspection

In the event of a hard landing, perform the “Annual Inspection” tasks prior to returning the aircraft to service.

Helicopter S/N: _____ Registration: _____
 Facility: _____ W.O.: _____
 A/F Total time: _____ Rin: _____
 Engine Total time: _____ Cycle: _____
 Date started: _____ Date completed: _____
 Signature: _____ License number: _____

Annual Inspection

Data Reference	Inspection Task Description	Mechanic
ETM1000 User Manual v7.0 or later	<p>Replacement of the Lithium Backup AA battery for Rev Q LRU's</p> <p>The new 3.6V AA Lithium “button top” style battery should be replaced bi-annually in order to provide continued retention of the date / time clock. The 3.6V AA Lithium battery is a standard industry lithium battery that can be purchased from AKV, Inc.</p> <p>Caution: Do not use a standard AA battery which has insufficient voltage of 1.5V</p> <p>With the A/C battery switch “OFF”, remove the battery holder cap by twisting counter clockwise. After the new battery has been installed, make sure the cap is installed and is fully tightened.</p> <p>You will now need to reset the date and time. Locate and connect the supplied RS232 cable to the “config” port just below the J1 connector. Connect the cable to a PC running Microsoft XP, 2000, Vista or Windows 7 or 10 using the 9 pin serial port. If a 9 pin serial port is not available then use the supplied USB / Serial adapter.</p>  <p>Replace with a new battery making sure the “+” side of the battery is down.</p>	

Data Reference	Inspection Task Description	Mechanic
ETM1000 User Manual v7.0 or later	<p>Operational Check</p> <p>The PC you will use must have been set up for communication via “HyperTerminal” prior to using the following procedure.</p> <p>Perform the following steps to verify proper operation of the ETM1000.</p> <ol style="list-style-type: none"> Use a PC running Windows XP, 2000, Vista or Windows 7/8 and connect to the “Config” port using the AKV supplied RS232 9-pin serial cable. Use the AKV supplied USB/Serial adaptor if your PC does not have a 9 pin serial port available. <p><u>If you have a -2 configuration without the pushbutton / indicators and audio side-tone, goto step 7</u></p> <ol style="list-style-type: none"> If the indicators are known to be activated from within the configuration settings of Hyperterminal, turn the A/C battery switch “ON” and observe that the annunciator lights are illuminated as follows during system BIT self-check: <div style="text-align: center; margin: 10px 0;">  </div> NOTE: If the WARNING light stays illuminated then there has been a prior exceedence that has not been acknowledge and cleared by maintenance. Remove the SD card to confirm what has been exceeded, verify corrective action then reset the light. If the SD light stays illuminated then check the SD card is inserted. If it is flashing then the SD card nearly full and data must be removed. If the SD card is bad then replace it. If the ERR light stays illuminated then there is a faulty connection with one of the signals to the ETM1000. Use HyperTerminal to determine which signal is faulty. * Ref. Appendix D for the Wiring Diagrams. If the audio is known to be activated from within the configuration settings of Hyperterminal and with the headset on your head, check that the audio side tone is heard by cycling the ETM circuit breaker. If it is weak or load, adjust the audio control head unit for a comforatbel level. It is not adjustable within the ETM1000. Check the cooling fan on the rear of the ETM1000 enclosure below the instrument panel for positive airflow by placing your hand next to the fan. Check the fan bearings are not noisy and the fan is clean and provides airflow by placing your hand over the rear of the ETM1000 enclosure. 	

Data Reference	Inspection Task Description	Mechanic
ETM1000 User Manual v7.0 or later	<ol style="list-style-type: none"> 9. On your PC, locate and run the Windows native program called "HyperTerminal" 10. Check that the Date and Time are correct. If not, then enter "S" for settings, enter your password and change the appropriate field as required. 11. Check that the Collective Time, Engine Run Time and Engine Starts are correct, If not then enter "S" for settings, enter the password and change the appropriate field as required. 12. Check that the OAT value in "HyperTerminal" is with 5 deg C of the A/C OAT probe. Keep in mind that the ETM1000 OAT probe is mounted on the belly of the A/C and could be receiving additional heat form the ground when compared to the A/C OAT probe mounted above the canopy. NOTE: If the cycle counting option is installed, then include the cycle Counting functional check described on the next page along with item 13, 14 & 15 below. 13. Start the A/C and run at ground idle. 14. Check that the engine parameters for Tq, MGT, N1, N2 and Nr values in "HyperTerminal" correspond with the indicators on the instrument panel. NOTE: During startup it is normal to sometimes see the ERR light flash on for 1-2 secs. This can vary between different A/C. 15. Run the A/C at flight speed and again, check the engine parameters for accuracy. 16. Return to ground idle and shutdown the engine. 17. Set the altimeter to 29.92 and verify the Pressure Altitude in "HyperTerminal" is within approximately 300ft. 18. Connect a pitot test set to the A/C pitot tube and while monitoring "HyperTerminal" check that the "Air speed Switch" goes from "0" (OFF) to a "1" (ON) when the test set is adjusted to 40Kt or greater. Tolerances - Activation +0 / -5 MPH and Deactivation -2 / -6 MPH * Ref. Appendix D for the Wiring Diagrams. <p>Coninued next page</p>	

Data Reference	Inspection Task Description	Mechanic
<p>ETM1000 Arriel 1 or LTS101 Cycle Counting Operating Specification</p>	<p>Cycle Counting Functional Check (installed as an option)</p> <p>The ETM1000 cycle counting option utilizes the same methodology as the SAFRAN validated stand-alone AKV cycle counter P/N 350NGTEC software called "Linear Ver Rev 1" or later approved revisions. The "Linear" version software records a K1 penalty for each 0.1% of speed increase due to the higher resolution capability of the ETM1000 cycle count system.</p> <p>NOTE: Cycle counting is available as an option only on the -1 configuration via the AKV remote display or displayed on the 3rd party Flightcell DZMx as shown in Fig. 3 & 4 above. In either case, the ETM1000 is the source for the cycle counting data. When using the AKV remote display the ETM1000 LRU must be Rev P or greater.</p> <p>1. Perform a ground run of the A/C and observe the first data screen for the Ng and Np speeds. These values are compared to the instrument panel indicators and should be within +/- 0.5%.</p> <p>Note: Ground runs of the A/C do not require cycle penalties to be recorded with the Arriel 1. A flight must be initiated in order to observe cycle penalties next to the "Fit" indication on the remote display.</p> <p>2. Perform a normal departure, cruise and landing. Note the Max Ng (K1) to the nearest 0.1% during takeoff and the min Ng speed (K2) during power reduction as well as how many times Ng was > 85% then < 85% then > 85%. After shutdown, compare your estimated cycle count using the Arriel Operating Specification document examples starting on Pg 11 for both Ng and Np.</p> <p>Make sure there is no significant discrepancy between the estimated values and recorded values shown on the cycle counter. If a discrepancy exists, refer to the troubleshooting sect. 7 or contact AKV for further assistance. In place of the ETM1000 cycle counter system which includes the ETM1000 LRU and possibly the remote display (if bad), continue operating the A/C using the SAFRAN pilot "manual cycle counting" procedure during removal and repair.</p> <p>After the ETM1000 LRU / remote display is repaired, repeat this "Annual Inspection" procedure to verify correct operation. Use the ETM1000 programming interface to update the cycle counting function with any new totals.</p> <p>END OF PROCEDURE</p>	

6. REMOVING PARTS

Removal instructions for the AKV ETM1000 and cycle counter remote display are listed below.

6.1 To Remove the ETM1000 LRU

Note: The ETM1000 Enclosure is secured to the mounting plate with four (4) ball studs that allow for a quick-release.

- a) Disconnect the airframe P1 electrical connector from the J1 ETM1000 enclosure electrical connector
- b) Place your hands on the front and rear of the enclosure
- c) Pull the enclosure directly towards you and away from the mounting bracket

6.2 To Remove the Airspeed Switch

See Appendix B for the Airspeed Switch Installation Instructions.

6.3 To Remove the Pushbutton Switches

See Appendix C for the Pushbutton Indicator Removal / Installation Instructions.

6.4 To Remove the (optional) Cycle Counting Remote Display

- d) Disconnect the airframe electrical connector from the cycle counter enclosure
- e) Remove the 2 mounting screws holding the cycle counter to the bracket
- f) Install a plastic bag over the airframe connector and secure with zip ties.

NOTE: To Remove the 3rd Party Flightcell DZMx, refer to the OEM Flightcell DZMx manual

7. TROUBLESHOOTING

The troubleshooting table provides the mechanic with guidance for diagnosing malfunctions and the recommended course of action to remedy the fault. Additional help can be found in the Inspection Task (operational task) starting on Pg. 10 or by contacting AKV. See Page 1 for support contact information.

ETM1000

Fault	Probable Cause	Action
No power to unit when aircraft battery power applied	1. Check Circuit breaker 2. Check J1 Connector	1. Reset the Circuit breaker 2. Turn off aircraft battery power and check connector with reference to Appendix D Drawings
Minimal or no airflow from cooling fan.	Cooling fan inoperative.	Check for obstruction and with the power off, check that the fan moves freely and is clean of dust.
Loss of date and time clock indicated by flashing ERR and confirmation with Hyperterminal connected	1. Battery hold cap loose 2. Battery is weak. Min 2.9V	1. Tighten cap 2. Replace battery with a new one

Cycle Counting (installed as an option)

Fault	Probable Cause	Action
No power to the remote display when aircraft battery power applied	1. Check ETM1000 Circuit breaker 2. Check J1 Connector	1. Reset the Circuit breaker 2. Turn off aircraft battery power and check connector with reference to Appendix D Drawings
High Ng and or Np cycles	Engine Signal Generator(s) noisy	Verify electrical connector at Signal Generator is clean and secure. ** <u>Perform signal quality check below.</u>
Blinking cursor in the top left corner of the remote display or no display at all	Electro-static discharge has damaged the LCD	Confirm that a ground bonding strap from the enclosure to airframe ground is present, secure and has < 3 milliohm resistance. Remote Display will need to be returned to AKV for repair. See Sect. 6 - Remove Parts

NOTE: If the above actions do not resolve the issue, contact AKV for additional assistance or to arrange for the unit to be returned for repair.

Signal quality check

1. Connect an oscilloscope to the ETM1000 airframe wire harness at the wiring splice point for Ng or Np behind the indicators for whichever signal is of concern.

Refer to attached Appendix D drawings ETM350-003 for N1/Ng and N2/Np connections

2. Ground run the A/C at idle and observe the quality of the sine-wave signal is as shown in Fig 1.
3. Whilst observing the oscilloscope, slowly increase the FCL to the flight position and observe that the sine-wave remains as shown in Fig 1 thru the speed change.

NOTE: A bad sine-wave signal will look similar to figure 2. If this is observed, clean the signal generator connector and/or replace the signal generator and perform the test again to confirm the quality of the signal is displayed like Fig 1.

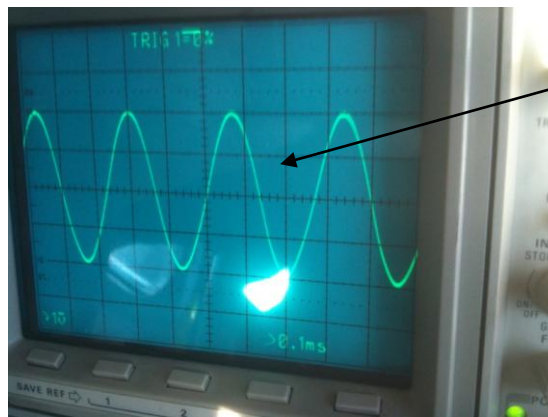


Fig. 1 (good)

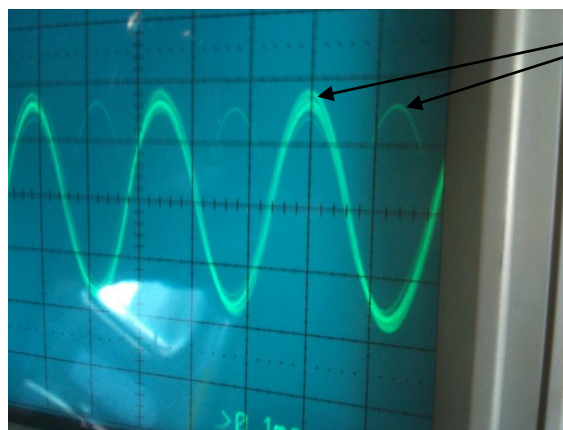


Fig. 2 (bad)

8. WEIGHT AND BALANCE CHANGES

In the event the ETM1000 has to be removed for maintenance, use the following table to adjust the aircraft weight and balance data.

Component	Weight
ETM1000 Enclosure P/N ETM350-006	1.25 Lbs 0.57 kg
Mounting Bracket P/N ETM350-007	0.25 Lbs 0.12 kg
(Optional) Cycle Counter Remote Display P/N ETMRD-004	0.25 Lbs 0.12 kg
(Optional) Cycle Counter Remote Display Bracket P/N AS350-ETMRD-001	0.75 Lbs 0.34 kg

Appendix A


A. CONFIGURING THE ETM1000

For Instructions on configuring the ETM1000 and optional cycle counting, refer to the ETM1000 User Manual v7.0 or later

Appendix B

B. AIRSPPEED SWITCH INSTALLATION INSTRUCTIONS

Intended Use:

The intended use of the airspeed switch is to provide an airspeed signal at ≥ 40 Knots for torque exceedence criteria. **NOTE:** The Airspeed switch has an approved provision as noted in the App. D of drawing ETM350-003 Sht 12. Ref  under the notes section to be electrically disconnected.

System Description:

This airspeed switch utilizes the existing Pitot system to provide the needed pressures for switch actuation. A ground leg is connected through the airspeed switch and will supply a ground connection for the ETM when the airspeed is ≥ 40 Knots.

Note: This installation will have negligible effect on aircraft weight and balance.

WARNING: When performing pitot/static systems certification you must connect the vent nipple on the AS9300-603-A switch to the static source on the test box to prevent damage to the switch diaphragm. The vent nipple bleed hole must be sealed to prevent leakage during testing. Do not connect vent nipple to the ships static system except during testing. Do not remove sealant compound from around switch terminals or pressure nipple bleed hole. Removal will cause leakage when performing a pitot system check.

CHECKOUT PROCEDURES:

Required Equipment:

- Pitot test set.

Test:

- 1) Test the Pitot system in accordance with the aircraft manufacturer's instructions. If the manufacturer has not issued instructions for testing Pitot systems, utilize the procedures contained in AC 43.13-1A or later approved revision.
- 2) Apply power to the aircraft and observe that the circuit protective device (circuit breaker or fuse) does not "trip" or "blow". Repair wiring if necessary for proper circuit protective device operation.
- 3) After the Pitot system has been successfully leak tested, adjust Pitot test set to indicate zero knots.
- 4) Observe that the Airspeed is "OFF" via the laptop PC running "HyperTerminal". If the Airspeed is "ON," check for the proper airspeed switch electrical connections. Incorrect switch connections will cause the ETM to not function correctly when determining a torque exceedence in flight.
- 5) Adjust the airspeed to equal 40 Knots or greater and observe that the Airspeed is "ON". If the Airspeed does not turn "ON", check connections on airspeed switch or repair wiring as necessary.

NOTE: Tolerances are Activation +0 / -5 MPH and Deactivation -2 / -6 MPH

- 6) Disconnect the Pitot test equipment from the aircraft.
- 7) Make the appropriate entries in the aircraft log book indicating a Pitot system test, and the airspeed switch installation have been completed.

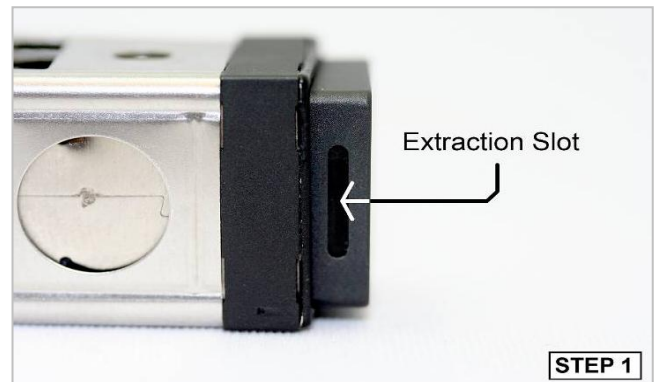
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Appendix C

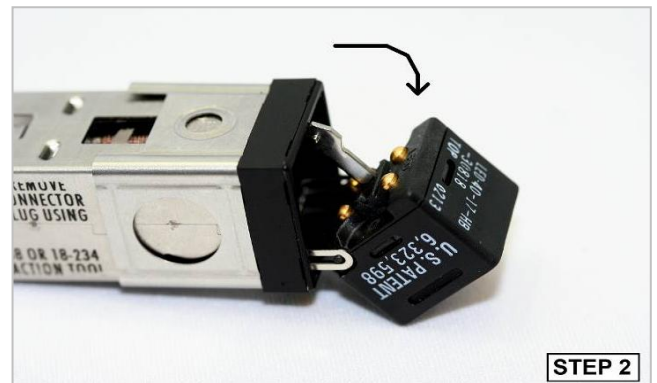
C. Pushbutton Indicator Removal/Installation Instructions

- 1) Examine the two sides of the switch cap to ensure the extraction slots are completely visible.

Note: To release the switch cap from the actuated position, simply push in the switch cap and allow the cap to return to the released position.



- 2) Extract the switch cap by using the Cap Extractor Tool (Part Number: 17-150) or by applying finger pressure on two sides of the switch cap then pull the switch cap from the switch body.



- 3) Remove switch cap from the switch body by gently removing the cap pins from the metallic retainer.



- 4) Remove the mounting sleeve by sliding the sleeve over the switch body from the back.

Note: The optional spacer can also be removed by sliding it from the back of the switch body.



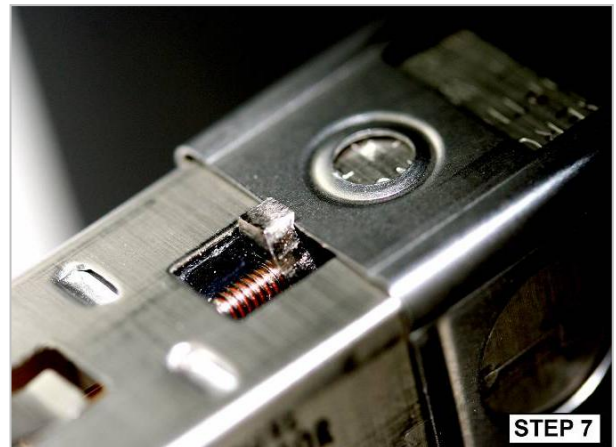
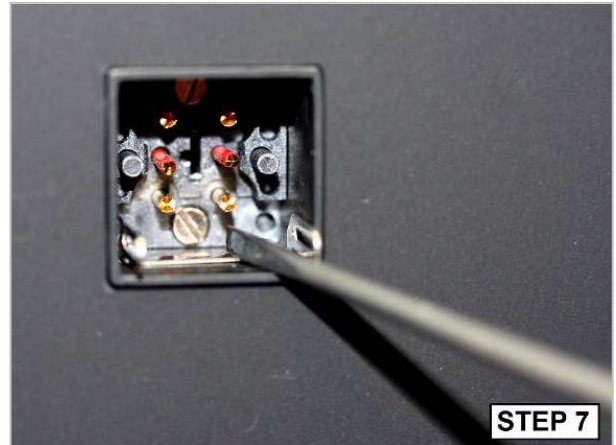
- 5) Insert the back of the switch body into the panel cutout and slide it through the panel from the front. Ensure the switch body label "TOP" is up.



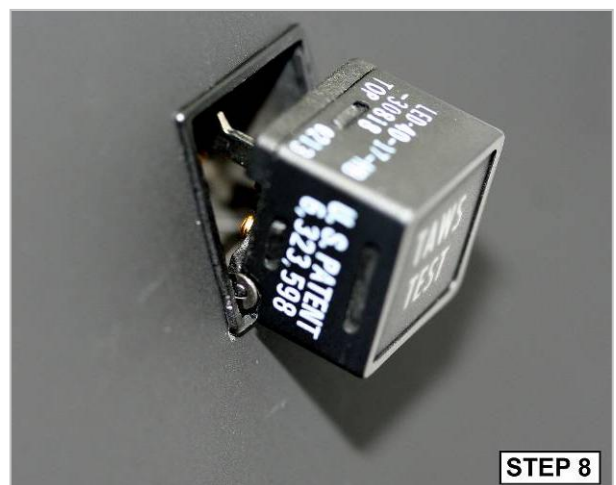
- 6) From behind the mounting panel, replace the mounting sleeve onto the switch body and slide it up to the back of the mounting panel.



- 7) From the front and inside of the switch body, tighten the two screws until the Integral Mounting Hardware pulls the mounting sleeve tight up against the mounting panel. Typical torque is 18 inch ounces.



- 8) Replace the switch cap in the switch body by inserting the cap pins into the metallic retainer and push the cap into the switch body.



Appendix D

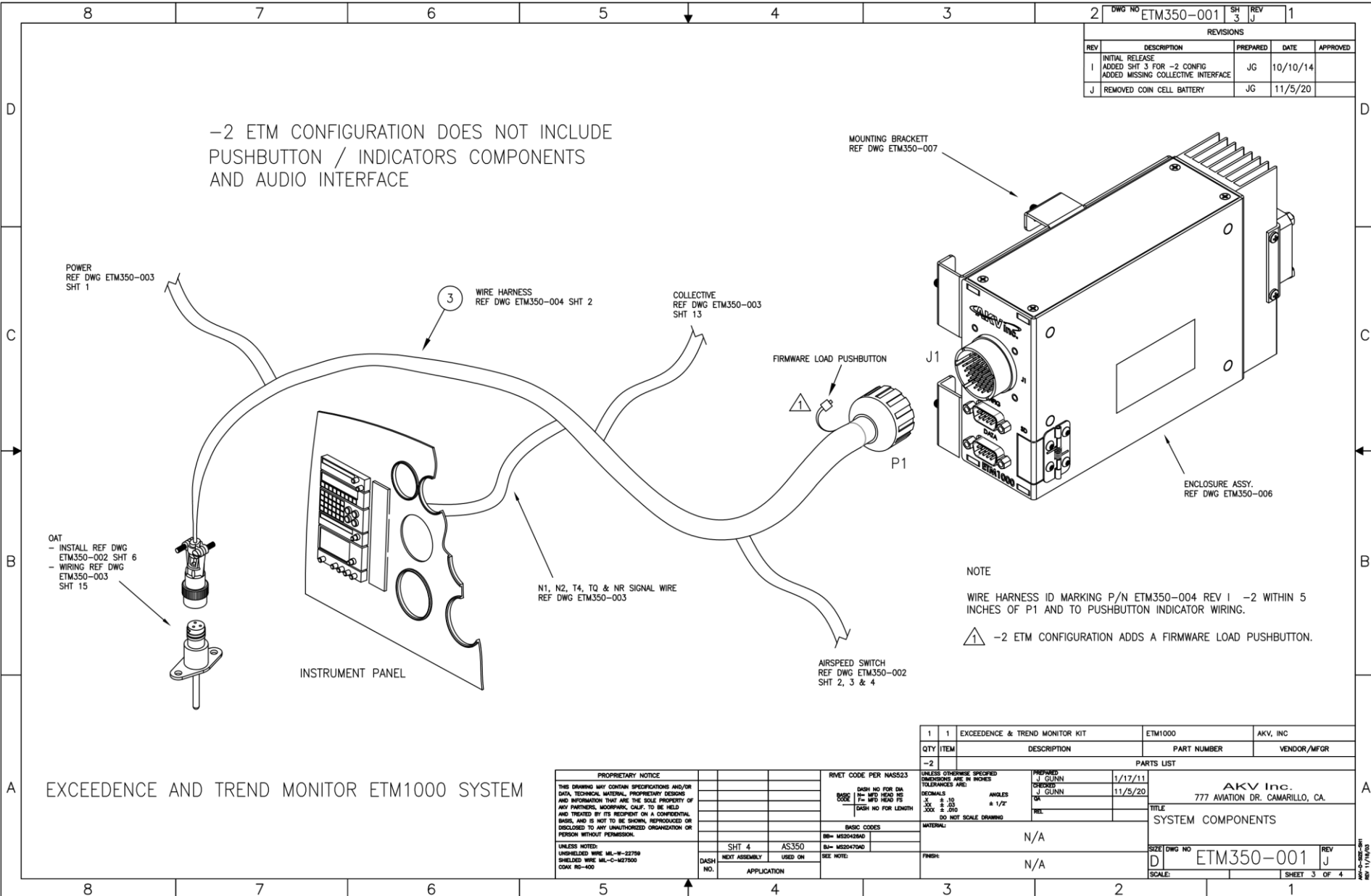
D. WIRING DIAGRAMS

This chart is a list of all MDL Rev R drawings found within Appendix D, in order of appearance.

DRAWING#	LEVEL	TITLE	REV	DATE
ETM350-001 Sht 1	SYSTEM COMPONENTS	TOP ASSY	K	11/5/20
ETM350-001 Sht 2	SYSTEM COMPONENTS	-1 CONFIGURATION	J	11/5/20
ETM350-001 Sht 3	SYSTEM COMPONENTS	-2 CONFIGURATION	J	11/5/20
ETM350-001 Sht 4	SYSTEM COMPONENTS	CABLE ROUTING	I	10/10/14
ETM350-002 Sht 1	MECHANICAL INSTALL	LRU MOUNTING	H	11/5/20
ETM350-002 Sht 2	MECHANICAL INSTALL	AIRSPEED SWITCH	D	12/10/11
ETM350-002 Sht 3	MECHANICAL INSTALL	AIRSPEED SWITCH	D	12/10/11
ETM350-002 Sht 4	MECHANICAL INSTALL	AIRSPEED SWITCH	D	12/10/11
ETM350-002 Sht 5	MECHANICAL INSTALL	PUSHBUTTON / INDICATOR	I	10/10/14
ETM350-002 Sht 6	MECHANICAL INSTALL	OAT SENSOR	E	11/5/20
ETM350-003 Sht 1	WIRING INSTALLATION	POWER	NC	1/17/11
ETM350-003 Sht 2	WIRING INSTALLATION	ARRIEL 1 THERMOUCOUPLE	H	7/1/20
ETM350-003 Sht 3	WIRING INSTALLATION	LTS101 THERMOUCOUPLE	A	7/1/20
ETM350-003 Sht 4	WIRING INSTALLATION	ARRIEL 1 Tq	H	1/29/14
ETM350-003 Sht 5	WIRING INSTALLATION	LTS101 Tq	J	12/19/24
ETM350-003 Sht 6	WIRING INSTALLATION	ARRIEL 1 N1 / Ng	G	12/29/12
ETM350-003 Sht 7	WIRING INSTALLATION	LTS101 N1 / Ng	NC	1/17/11
ETM350-003 Sht 8	WIRING INSTALLATION	ARRIEL 1 N2 / Np	G	12/29/12
ETM350-003 Sht 9	WIRING INSTALLATION	LTS101 N2 / Np	NC	1/17/11
ETM350-003 Sht 10	WIRING INSTALLATION	ARRIEL 1 Nr	H	1/29/14
ETM350-003 Sht 11	WIRING INSTALLATION	LTS101 Nr	H	1/29/14
ETM350-003 Sht 12	WIRING INSTALLATION	AIRSPEED SWITCH	B	1/1/25
ETM350-003 Sht 13	WIRING INSTALLATION	COLLECTIVE SWITCH	A	7/24/17
ETM350-003 Sht 14	WIRING INSTALLATION	AUDIO PANEL INTERFACE	I	10/10/14
ETM350-003 Sht 15	WIRING INSTALLATION	OAT SENSOR	H	11/5/20
ETM350-003 Sht 16	WIRING INSTALLATION	PUSHBUTTON / INDICATOR	I	10/10/14
ETM350-003 Sht 17	WIRING INSTALLATION	LOAD WEIGHT (HOOK) INTERFACE	NC	1/1/25
ETM350-004 Sht 1	WIRING INSTALLATION	-1 WIRE HARNESS	K	1/1/25
ETM350-004 Sht 2	WIRING INSTALLATION	-2 WIRE HARNESS	K	1/1/25
ETM350-005 Sht 3	WIRING INSTALLATION	J1 PINOUT	J	1/1/25
ETM350-010 Sht 1	WIRING INSTALLATION	CONFIGURATION PORT	A	3/6/18
ETM350-011 Sht 1	WIRING INSTALLATION	DATA PORT	B	1/1/25
AS350-ETMRD-002 Sht 1	WIRING INSTALLATION	REMOTE DISPLAY INSTALLATION	NC	3/6/18
ETMRD-003 Sht 1	WIRING INSTALLATION	REMOTE DISPLAY WIRING	A	3/5/26
ETMRD-003 Sht 2	WIRING INSTALLATION	REMOTE DISPLAY WIRING	NC	3/5/26

REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
I	INITIAL RELEASE			
I	ADDED SHT 3 FOR -2 CONFIG	JG	10/10/14	
J	ADDED MISSING COLLECTIVE INTERFACE			
J	REMOVED COIN CELL BATTERY	JG	11/5/20	

-2 ETM CONFIGURATION DOES NOT INCLUDE
 PUSHBUTTON / INDICATORS COMPONENTS
 AND AUDIO INTERFACE

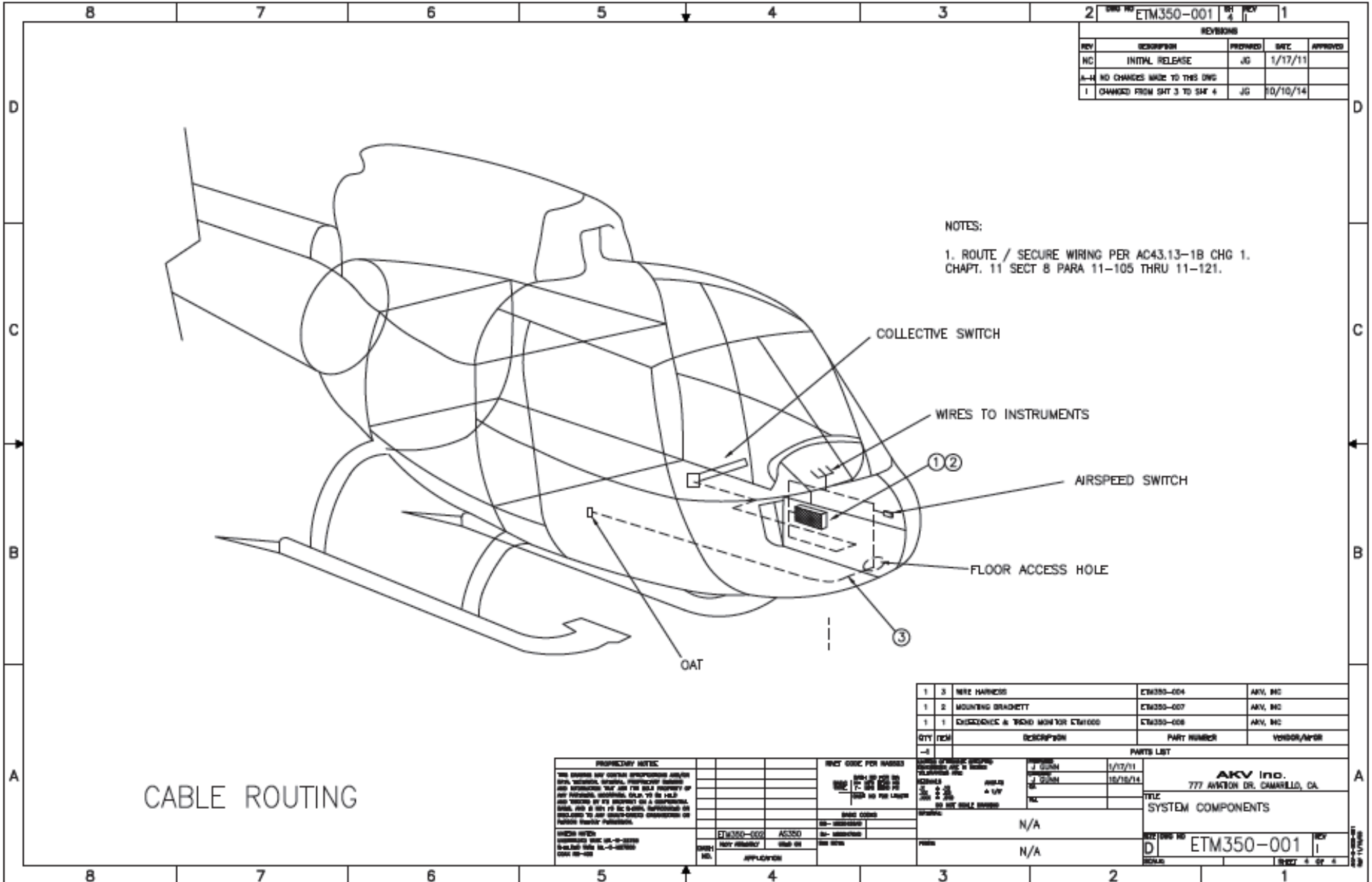


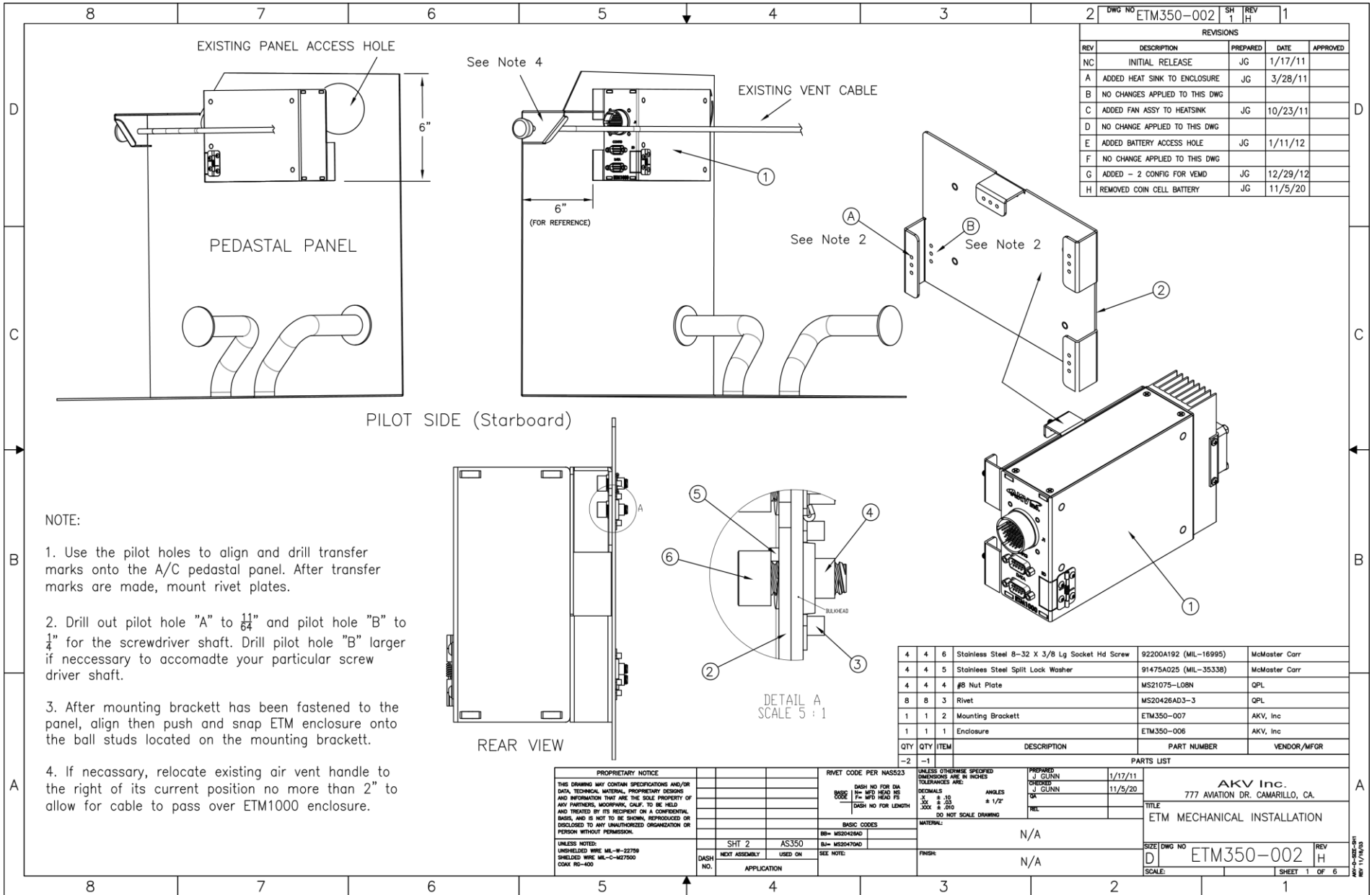
NOTE
 WIRE HARNESS ID MARKING P/N ETM350-004 REV 1 -2 WITHIN 5 INCHES OF P1 AND TO PUSHBUTTON INDICATOR WIRING.
 ⚠ -2 ETM CONFIGURATION ADDS A FIRMWARE LOAD PUSHBUTTON.

EXCEEDENCE AND TREND MONITOR ETM1000 SYSTEM

PROPRIETARY NOTICE THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGNS AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOOREAVIC, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.		UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE: DECIMALS .10 ANGLES ± 1/2° FRACTIONS .005 ± .010 DO NOT SCALE DRAWING	
UNLESS NOTED: UNSHELDED WIRE MIL-N-22759 SHIELDED WIRE MIL-C-427500 CONV RC-400		BASIC CODES SHT 4 AS350 DASH NO. APPLICATION	

1	1	EXCEEDENCE & TREND MONITOR KIT	ETM1000	AKV, INC
QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFR
	-2			
PARTS LIST				
PREPARED	J GUNN	1/17/11	AKV Inc.	
CHECKED	J GUNN	11/5/20	777 AVIATION DR. CAMARILLO, CA.	
TITLE				
SYSTEM COMPONENTS				
SIZE	DWG NO	REV	DATE	
D	ETM350-001	J	11/16/20	
SCALE:		SHEET	3	OF 4





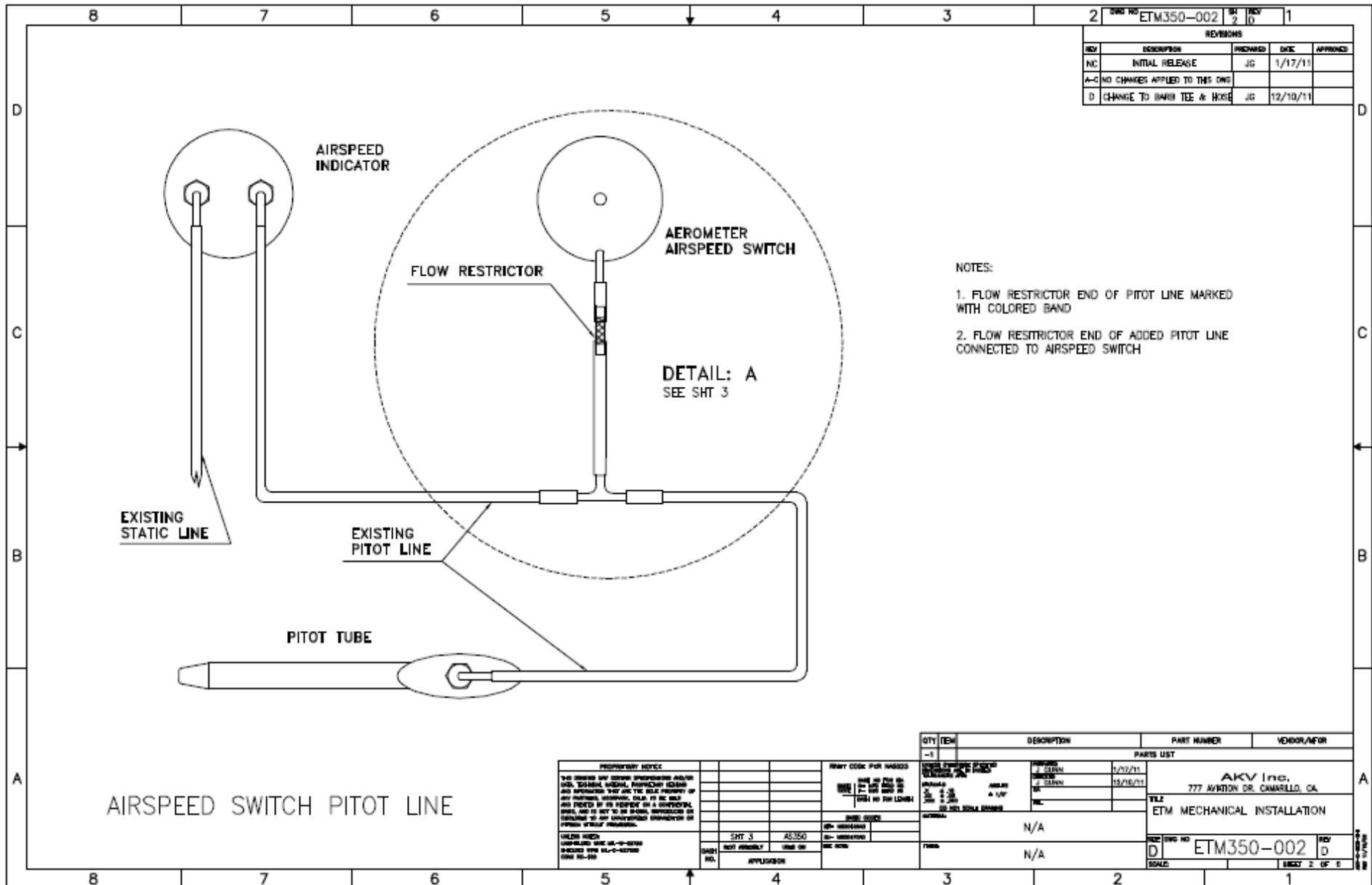
NOTE:

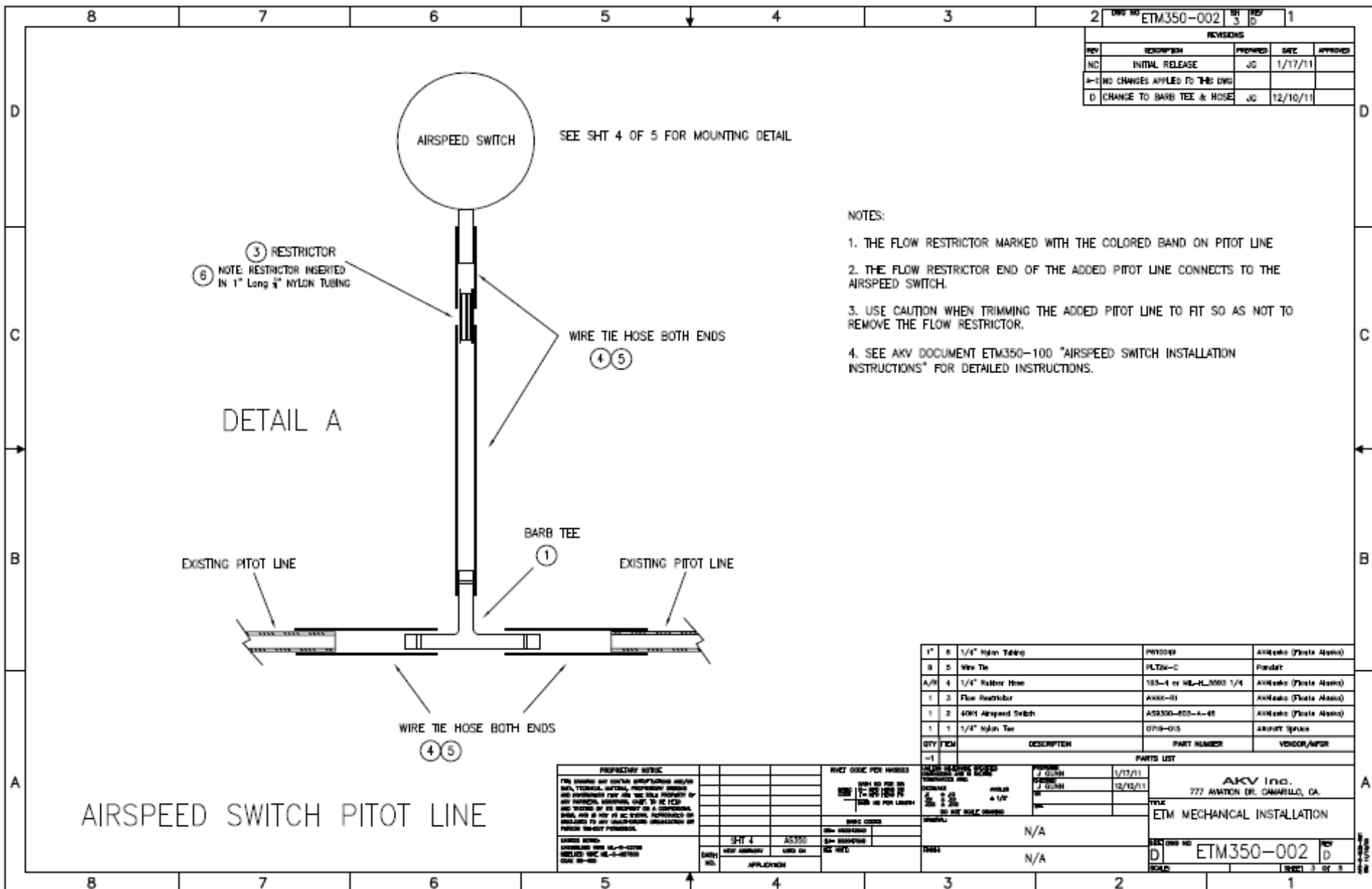
1. Use the pilot holes to align and drill transfer marks onto the A/C pedestal panel. After transfer marks are made, mount rivet plates.
2. Drill out pilot hole "A" to $\frac{11}{64}$ " and pilot hole "B" to $\frac{1}{4}$ " for the screwdriver shaft. Drill pilot hole "B" larger if necessary to accomadte your particular screw driver shaft.
3. After mounting brackett has been fastened to the panel, align then push and snap ETM enclosure onto the ball studs located on the mounting brackett.
4. If necessary, relocate existing air vent handle to the right of its current position no more than 2" to allow for cable to pass over ETM1000 enclosure.

REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A	ADDED HEAT SINK TO ENCLOSURE	JG	3/28/11	
B	NO CHANGES APPLIED TO THIS DWG			
C	ADDED FAN ASSY TO HEATSINK	JG	10/23/11	
D	NO CHANGE APPLIED TO THIS DWG			
E	ADDED BATTERY ACCESS HOLE	JG	1/11/12	
F	NO CHANGE APPLIED TO THIS DWG			
G	ADDED - 2 CONFIG FOR VEMD	JG	12/29/12	
H	REMOVED COIN CELL BATTERY	JG	11/5/20	

QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFR
4	4	6	Stainless Steel 8-32 X 3/8 Lg Socket Hd Screw	92200A192 (MIL-16995) McMaster Carr
4	4	5	Stainless Steel Split Lock Washer	91475A025 (MIL-35338) McMaster Carr
4	4	4	#8 Nut Plate	MS21075-LOBN OPL
8	8	3	Rivet	MS20426AD3-3 OPL
1	1	2	Mounting Brackett	ETM350-007 AKV, Inc
1	1	1	Enclosure	ETM350-006 AKV, Inc

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UNLESS NOTED: UNSHIELDED WIRE MIL-B-22759 SHIELDED WIRE MIL-C-467500 COAK RS-400		BASIC CODES BU= MS20426AD BJ= MS20470AD SEE NOTE:		MATERIAL: N/A FINISH: N/A	
DASH NO. APPLICATION SHT 2 AS350 NEXT ASSEMBLY USED ON		PREPARED J GUNN 1/17/11 CHECKED J GUNN 11/5/20		AKV Inc. 777 AVIATION DR. CAMARILLO, CA. TITLE ETM MECHANICAL INSTALLATION	
SCALE:		SIZE/DWG NO D ETM350-002 REV H		SHEET 1 OF 6	





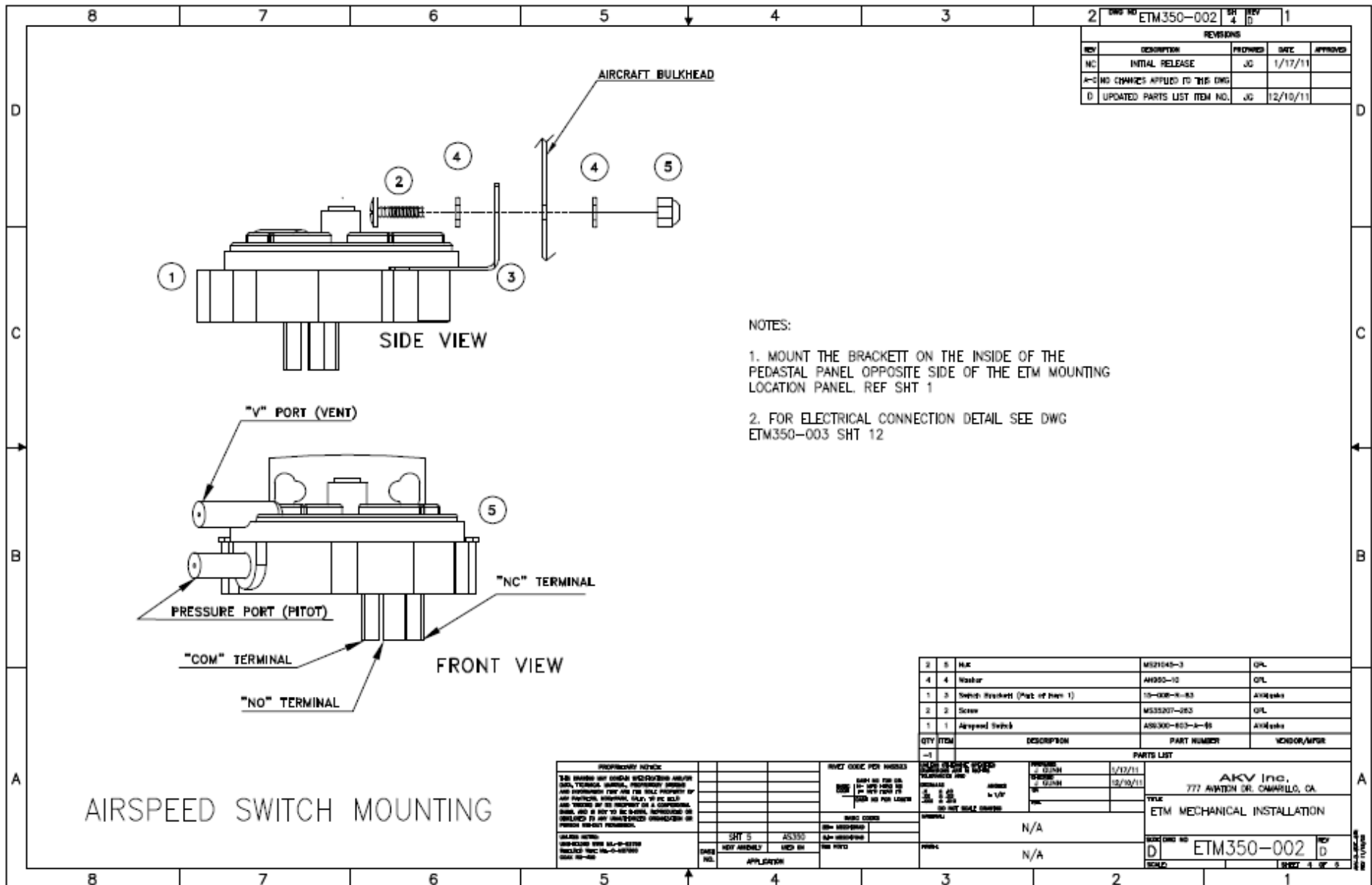
REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NO	INITIAL RELEASE	JG	1/17/11	
A-1	NO CHANGES APPLIED TO THIS DWG			
D	CHANGE TO BARB TEE & HOSE	JG	12/10/11	

- NOTES:
1. THE FLOW RESTRICTOR MARKED WITH THE COLORED BAND ON PITOT LINE
 2. THE FLOW RESTRICTOR END OF THE ADDED PITOT LINE CONNECTS TO THE AIRSPEED SWITCH.
 3. USE CAUTION WHEN TRIMMING THE ADDED PITOT LINE TO FIT SO AS NOT TO REMOVE THE FLOW RESTRICTOR.
 4. SEE AKV DOCUMENT ETM350-100 "AIRSPEED SWITCH INSTALLATION INSTRUCTIONS" FOR DETAILED INSTRUCTIONS.

QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/AMSR
1	6	1/4" Nylon Tubing	PW10049	AKWesco (Florida Airmen)
1	5	Wire Tie	PLTW-C	Fluoridel
A/R	4	1/4" Rubber Hose	183-4 or 184-4L-3000 1/4"	AKWesco (Florida Airmen)
1	3	Flow Restrictor	AKXX-01	AKWesco (Florida Airmen)
1	2	AKV Airspeed Switch	AS350-003-A-05	AKWesco (Florida Airmen)
1	1	1/4" Nylon Tee	0719-015	AKROUT Spares

PROPRIETARY NOTICE		REVISED CODES FOR AIRSPEED	
THIS DRAWING OR CONTENTS HEREOF IS UNCLASSIFIED EXCEPT WHERE SHOWN OTHERWISE AND IS THE PROPERTY OF AKV, INC. IT IS LOANED TO YOU BY AKV, INC. IT IS TO BE USED ONLY FOR THE PURPOSES SPECIFIED AND IS NOT TO BE REPRODUCED OR TRANSMITTED IN ANY FORM OR BY ANY MEANS, ELECTRONIC OR MECHANICAL, INCLUDING PHOTOCOPYING, RECORDING, OR BY ANY INFORMATION STORAGE AND RETRIEVAL SYSTEM, WITHOUT THE WRITTEN PERMISSION OF AKV, INC.		REVISED CODES FOR AIRSPEED	REVISED CODES FOR AIRSPEED
LARGER SIZES: CONTACT AKV FOR SIZES NOT LISTED HEREIN.		REVISED CODES FOR AIRSPEED	REVISED CODES FOR AIRSPEED
DATE	DESCRIPTION	REVISED CODES FOR AIRSPEED	REVISED CODES FOR AIRSPEED
	APPLICATION	REVISED CODES FOR AIRSPEED	REVISED CODES FOR AIRSPEED

AKV Inc.
 777 AVIATION DR. CARROLLTON, GA.
 TITLE: ETM MECHANICAL INSTALLATION
 DWG NO: ETM350-002
 REV: F
 DATE: 12/10/11



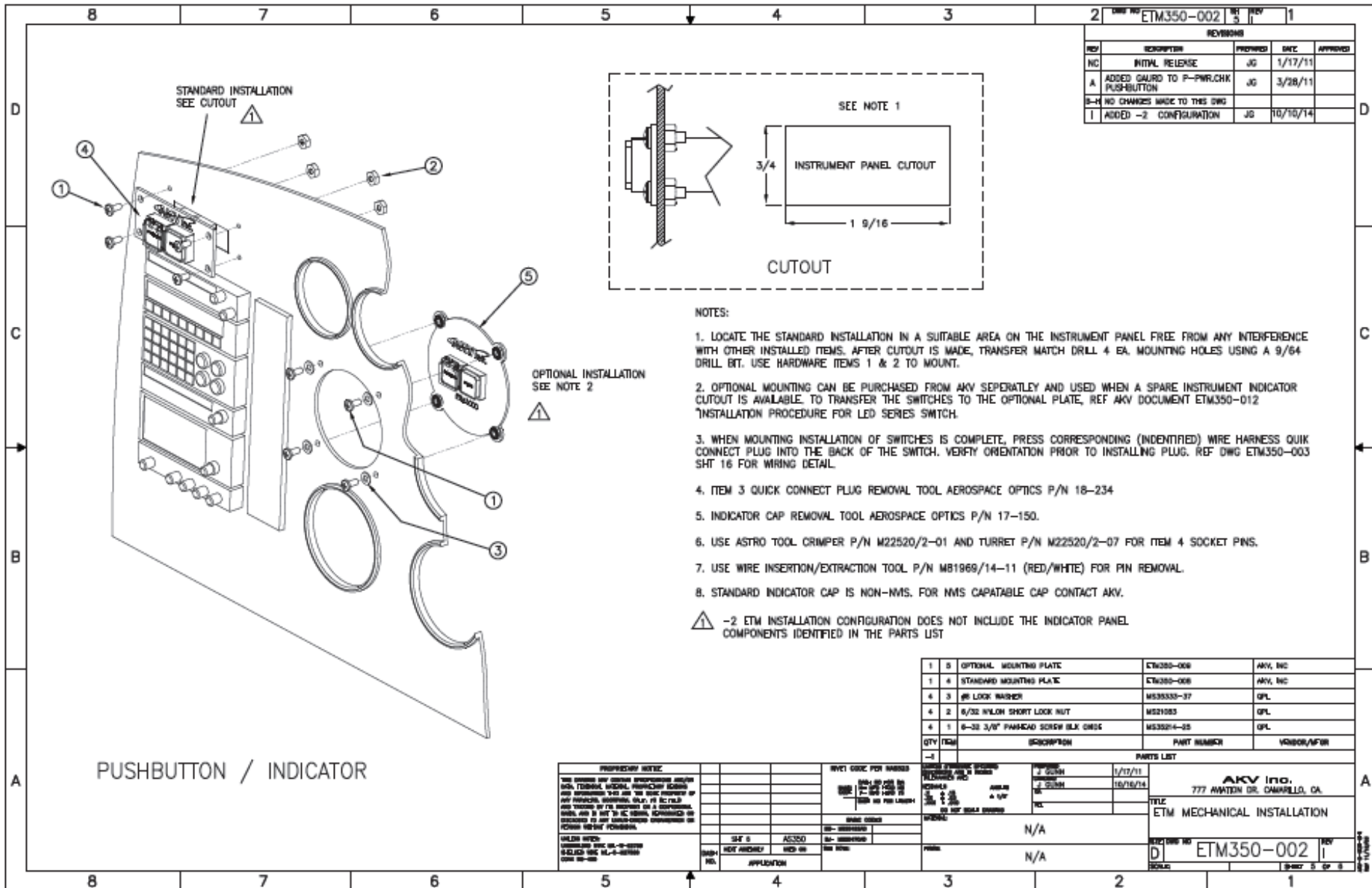
REV	DESCRIPTION	PROW'D	DATE	APPROV'D
NC	INITIAL RELEASE	JG	1/17/11	
A	NO CHANGES APPLIED TO THIS DWG			
D	UPDATED PARTS LIST ITEM NO.	JG	12/10/11	

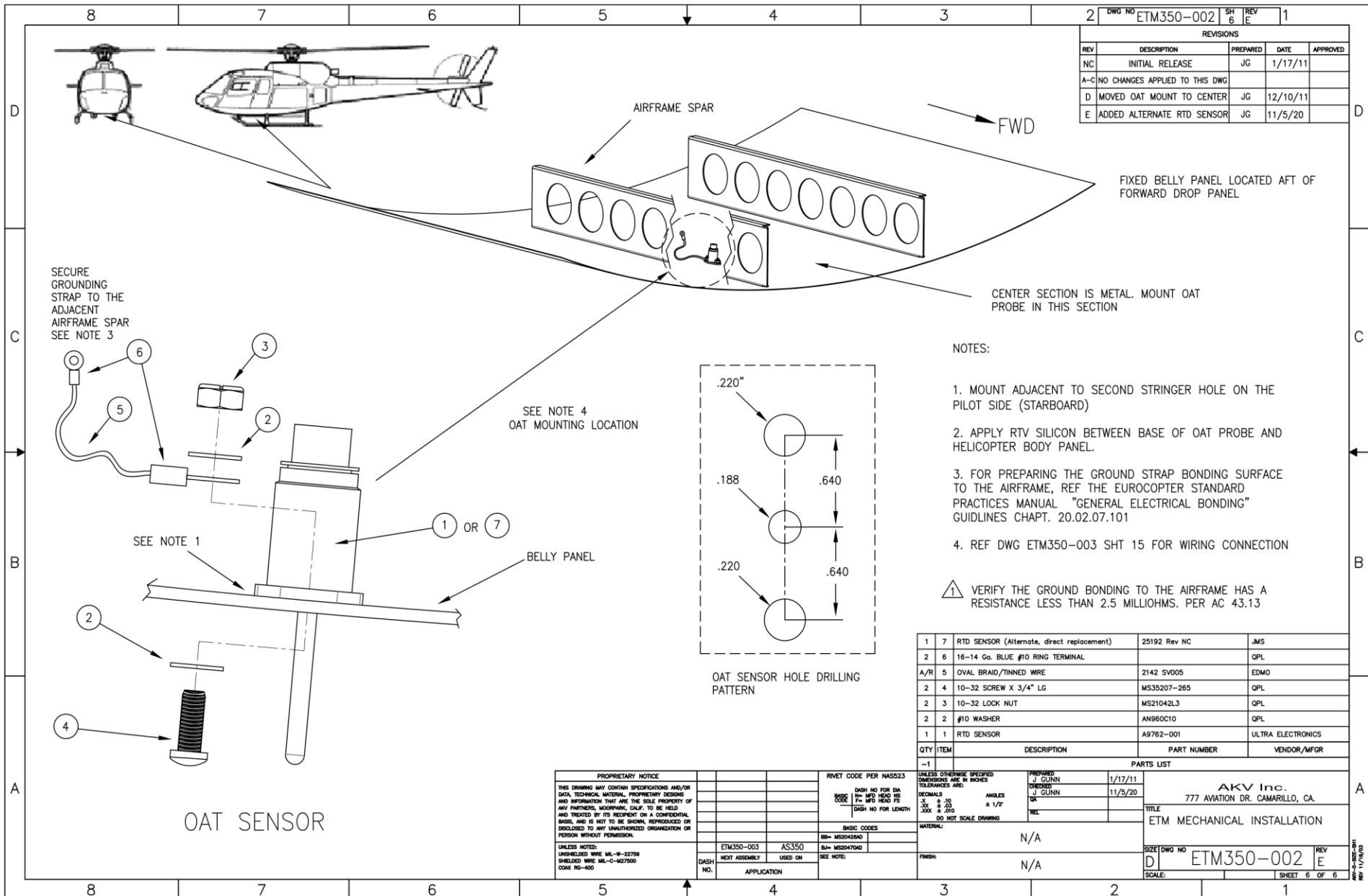
NOTES:

1. MOUNT THE BRACKET ON THE INSIDE OF THE PEDASTAL PANEL OPPOSITE SIDE OF THE ETM MOUNTING LOCATION PANEL. REF SHT 1
2. FOR ELECTRICAL CONNECTION DETAIL SEE DWG ETM350-003 SHT 12

QTY	ITEM	DESCRIPTION	PART NUMBER	UNIDOR/MPR
2	5	NUT	MS21045-3	QPL
4	4	Washer	AH950-12	QPL
1	3	Switch Bracket (Part of Item 1)	15-000-1-83	AKV/whs
2	2	Screw	MS3207-283	QPL
1	1	Airspeed Switch	AS3500-103-1-01	AKV/whs

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DATE: 12/10/11 BY: JG	SHT 5 AS350 NEXT ASSEMBLY USED BY	N/A	N/A	TITLE ETM MECHANICAL INSTALLATION
DWG NO: ETM350-002 REV: D	APPR: JG	N/A	N/A	DWG NO: ETM350-002 REV: D



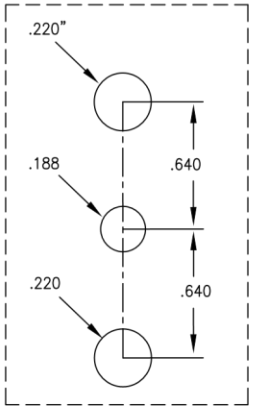


REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A-C	NO CHANGES APPLIED TO THIS DWG			
D	MOVED OAT MOUNT TO CENTER	JG	12/10/11	
E	ADDED ALTERNATE RTD SENSOR	JG	11/5/20	

NOTES:

1. MOUNT ADJACENT TO SECOND STRINGER HOLE ON THE PILOT SIDE (STARBOARD)
2. APPLY RTV SILICON BETWEEN BASE OF OAT PROBE AND HELICOPTER BODY PANEL.
3. FOR PREPARING THE GROUND STRAP BONDING SURFACE TO THE AIRFRAME, REF THE EUROCOPTER STANDARD PRACTICES MANUAL "GENERAL ELECTRICAL BONDING" GUIDELINES CHAPT. 20.02.07.101
4. REF DWG ETM350-003 SHT 15 FOR WIRING CONNECTION

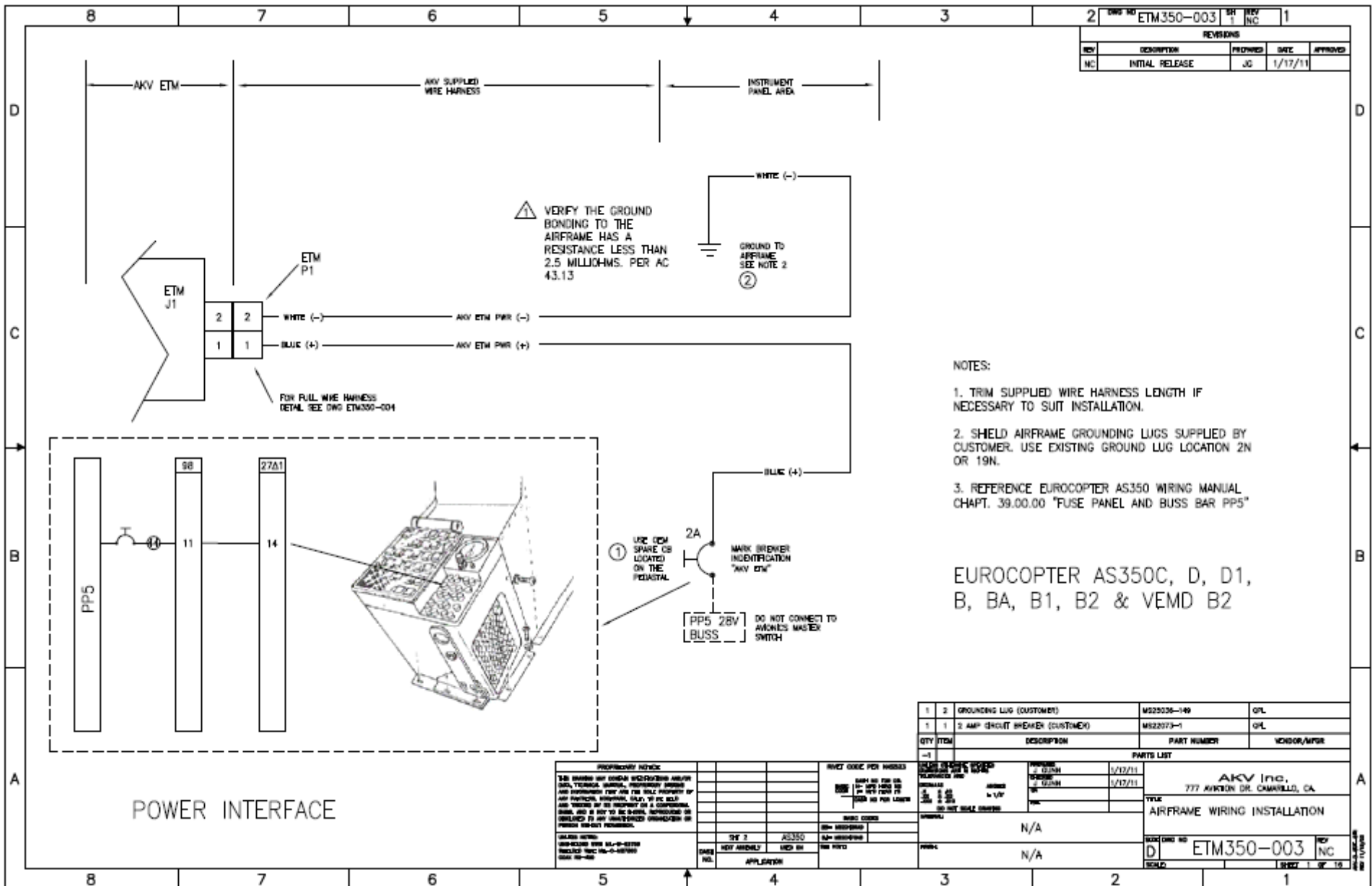
1 VERIFY THE GROUND BONDING TO THE AIRFRAME HAS A RESISTANCE LESS THAN 2.5 MILLIOHMS. PER AC 43.13

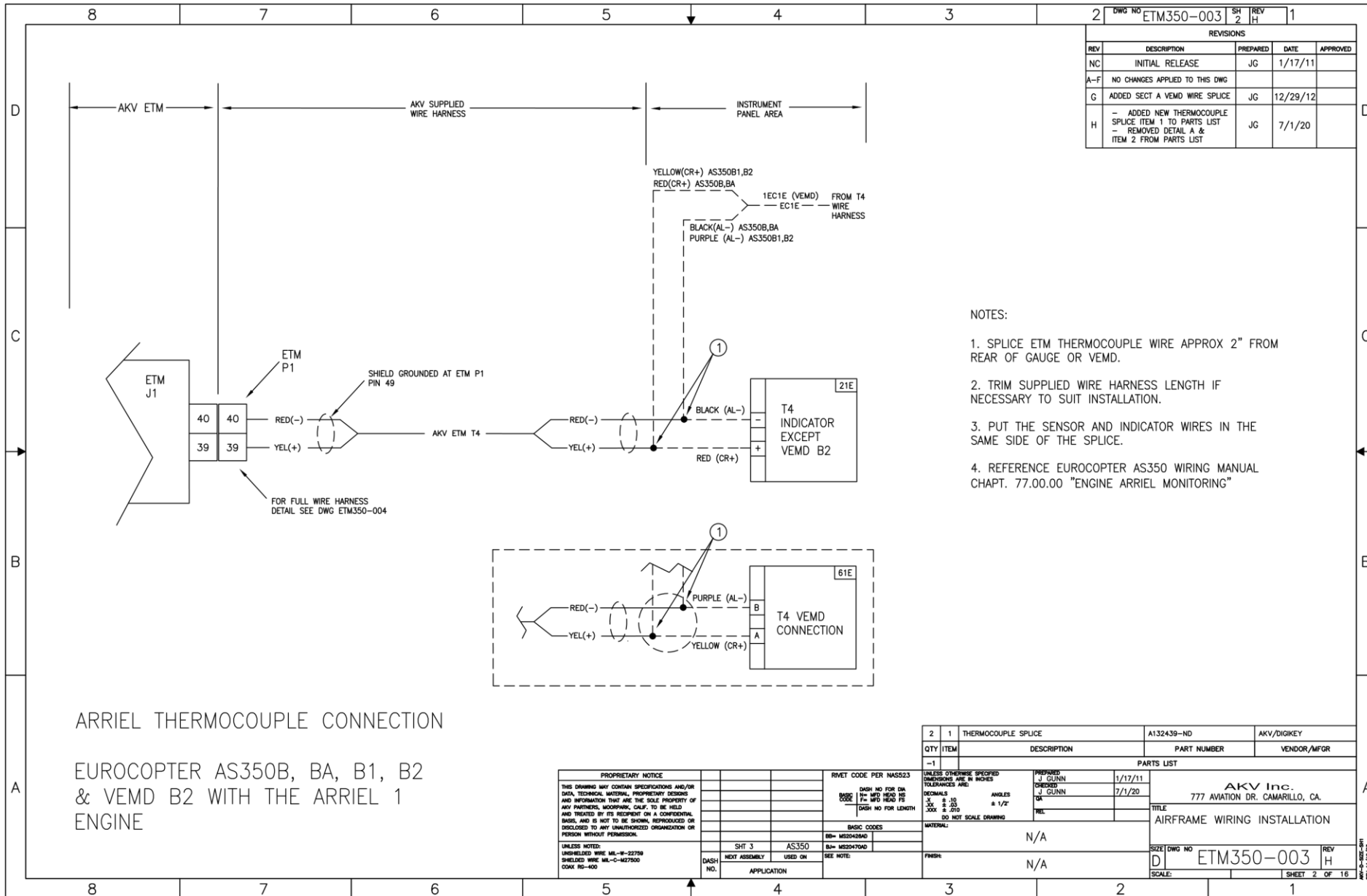


QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFR
1	7	RTD SENSOR (Alternate, direct replacement)	25192 Rev NC	JMS
2	6	16-14 Gg. BLUE #10 RING TERMINAL		GPL
A/R	5	OVAL BRAID/TINNED WIRE	2142 SV005	EDMO
2	4	10-32 SCREW X 3/4" LG	MS35207-285	GPL
2	3	10-32 LOCK NUT	MS21042L3	GPL
2	2	#10 WASHER	AN960C10	GPL
1	1	RTD SENSOR	A9762-001	ULTRA ELECTRONICS

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UNLESS NOTED: UNSHIELDED WIRE MIL-8-22750 SHIELDED WIRE MIL-C-427500 COAX RS-400		BASIC CODES BM= MS20480D BM= MS20470D SEE NOTE:	
DASH NO.	ETM350-003	AS350	USED ON
APPLICATION			

PREPARED J GUNN 1/17/11 CHECKED J GUNN 11/5/20 QA		PARTS LIST AKV Inc. 777 AVIATION DR. CAMARILLO, CA.	
MATERIAL: N/A		TITLE ETM MECHANICAL INSTALLATION	
FINISH: N/A		SIZE/DWG NO D ETM350-002 REV E	
SCALE:		SHEET 6 OF 6	





REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A-F	NO CHANGES APPLIED TO THIS DWG			
G	ADDED SECT A VEMD WIRE SPLICE	JG	12/29/12	
H	- ADDED NEW THERMOCOUPLE SPLICE ITEM 1 TO PARTS LIST - REMOVED DETAIL A & ITEM 2 FROM PARTS LIST	JG	7/1/20	

NOTES:

1. SPLICE ETM THERMOCOUPLE WIRE APPROX 2" FROM REAR OF GAUGE OR VEMD.
2. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATION.
3. PUT THE SENSOR AND INDICATOR WIRES IN THE SAME SIDE OF THE SPLICE.
4. REFERENCE EUROCOPTER AS350 WIRING MANUAL CHAPT. 77.00.00 "ENGINE ARRIEL MONITORING"

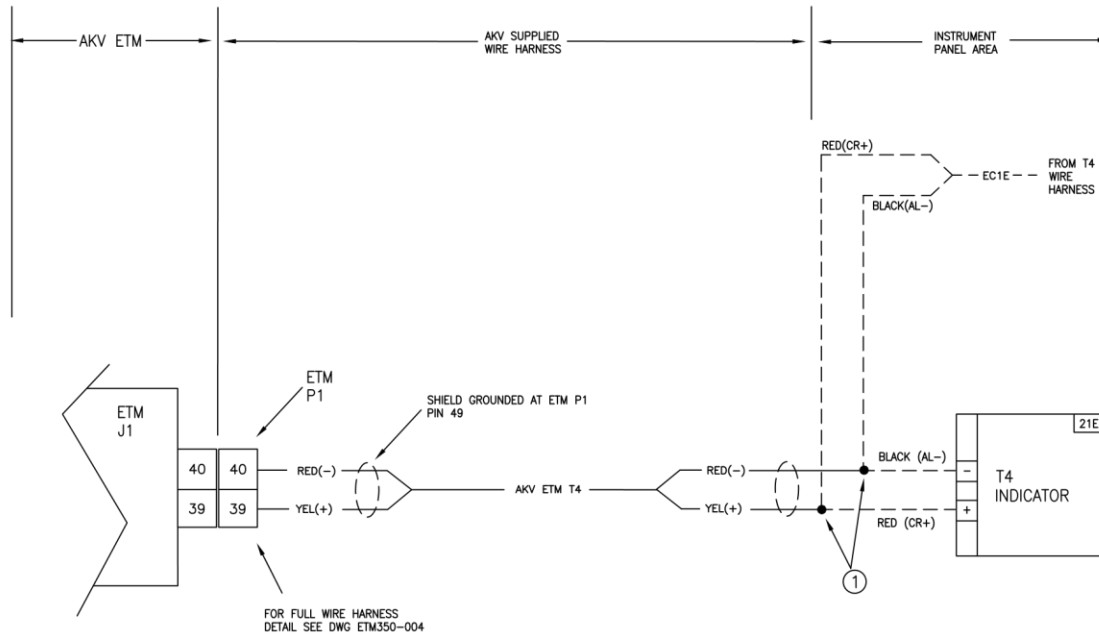
ARRIEL THERMOCOUPLE CONNECTION

EUROCOPTER AS350B, BA, B1, B2
 & VEMD B2 WITH THE ARRIEL 1
 ENGINE

PROPRIETARY NOTICE THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGNS AND INFORMATION THAT ARE THE SOLE PROPERTY OF ANY PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.		RIVET CODE PER NAS523 DASH NO FOR ON DASH NO FOR HEAD DASH NO FOR LENGTH		BASIC CODES SH= MS20454AD BL= MS20470AD SEE NOTE:	
UNLESS NOTED: UNSHIELDED WIRE MIL-W-22759 SHIELDED WIRE MIL-C-1427500 OXAK RG-400		SHT 3 AS350 NEXT ASSEMBLY USED ON APPLICATION		FINISH: N/A	

QTY	1	DESCRIPTION	THERMOCOUPLE SPLICE	A132439-ND	AKV/DIG/KEY
ITEM				PART NUMBER	VENDOR/MFR
PARTS LIST					
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE:		PREPARED	1/17/11	AKV Inc.	
DECIMALS		DESIGNED	7/1/20	777 AVIATION DR. CAMARILLO, CA	
ANGLES		CHK		TITLE	
DIA ± .015		REL		AIRFRAME WIRING INSTALLATION	
HOLE ± .010				SIZE (DWG NO)	
DO NOT SCALE DRAWING				D ETM350-003	
MATERIAL:				REV	
N/A				H	
SCALE:				SHEET 2 OF 16	

2		DWG NO	ETM350-003	SH	3	REV	A	1
REVISIONS								
REV	DESCRIPTION	PREPARED	DATE	APPROVED				
NC	INITIAL RELEASE	JG	1/17/11					
A	CHANGED PARTS LIST ITEM 1	JG	7/1/20					



NOTES:

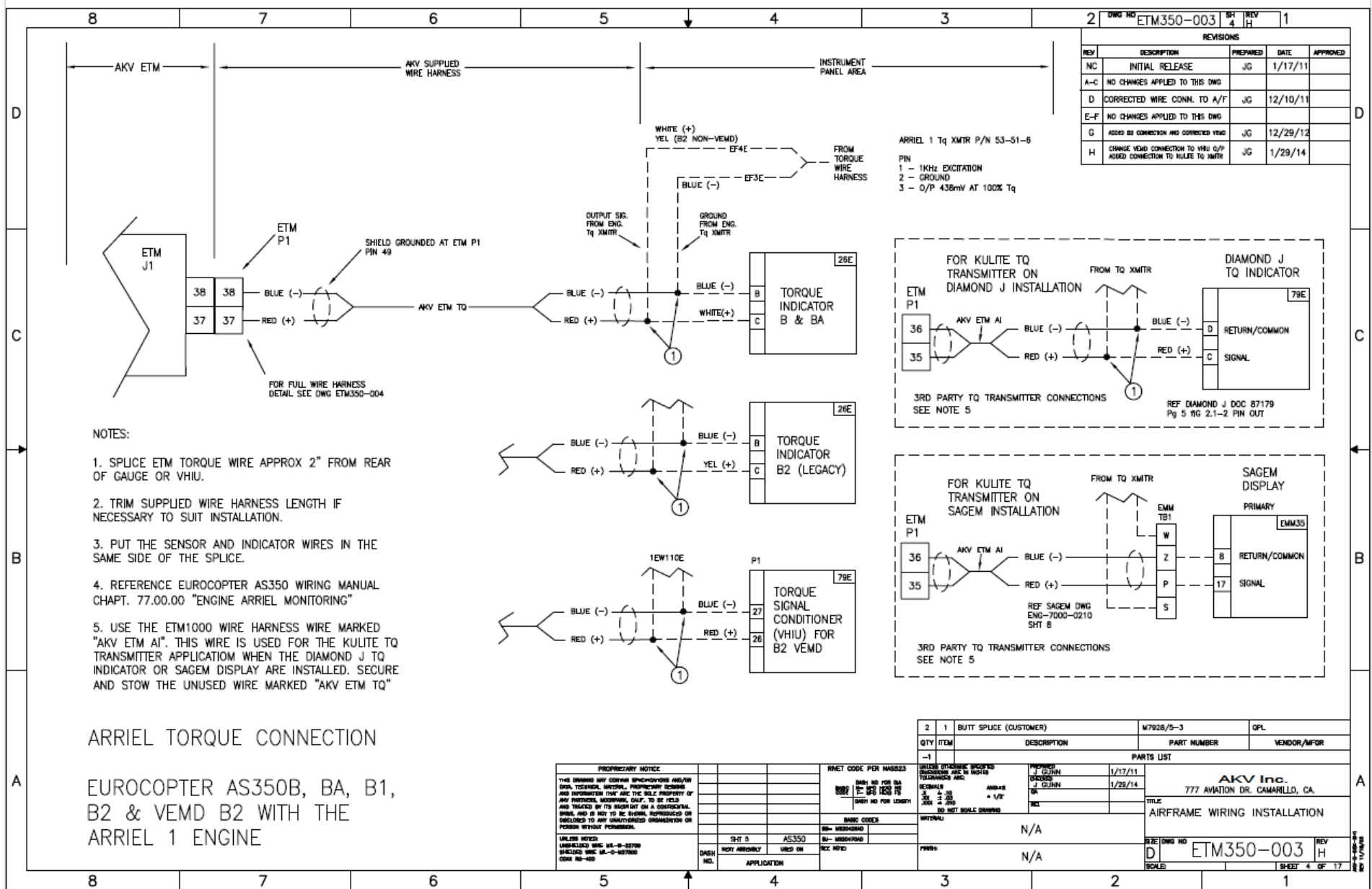
1. IF WORKING WITH LTS101 CONVERTED AS350 MODIFIED WITH THE SOLOY OR HELILYNX LTS101 STC ENGINE CONVERSIONS, REFER TO THE RESPECTIVE CONVERSION DOCUMENTATION DRAWINGS FOR THE ASSIGNED WIRE AND INDICATOR TERMINAL NUMBER.
2. SPLICE ETM THERMOCOUPLE WIRE APPROX 2" FROM REAR OF GAUGE.
3. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATION.
4. PUT THE SENSOR AND INDICATOR WIRES IN THE SAME SIDE OF THE SPLICE.
5. REFERENCE EUROCOPTER AS350 WIRING MANUAL CHAPT. 77.00.00 "ENGINE LYCOMING MONITORING"

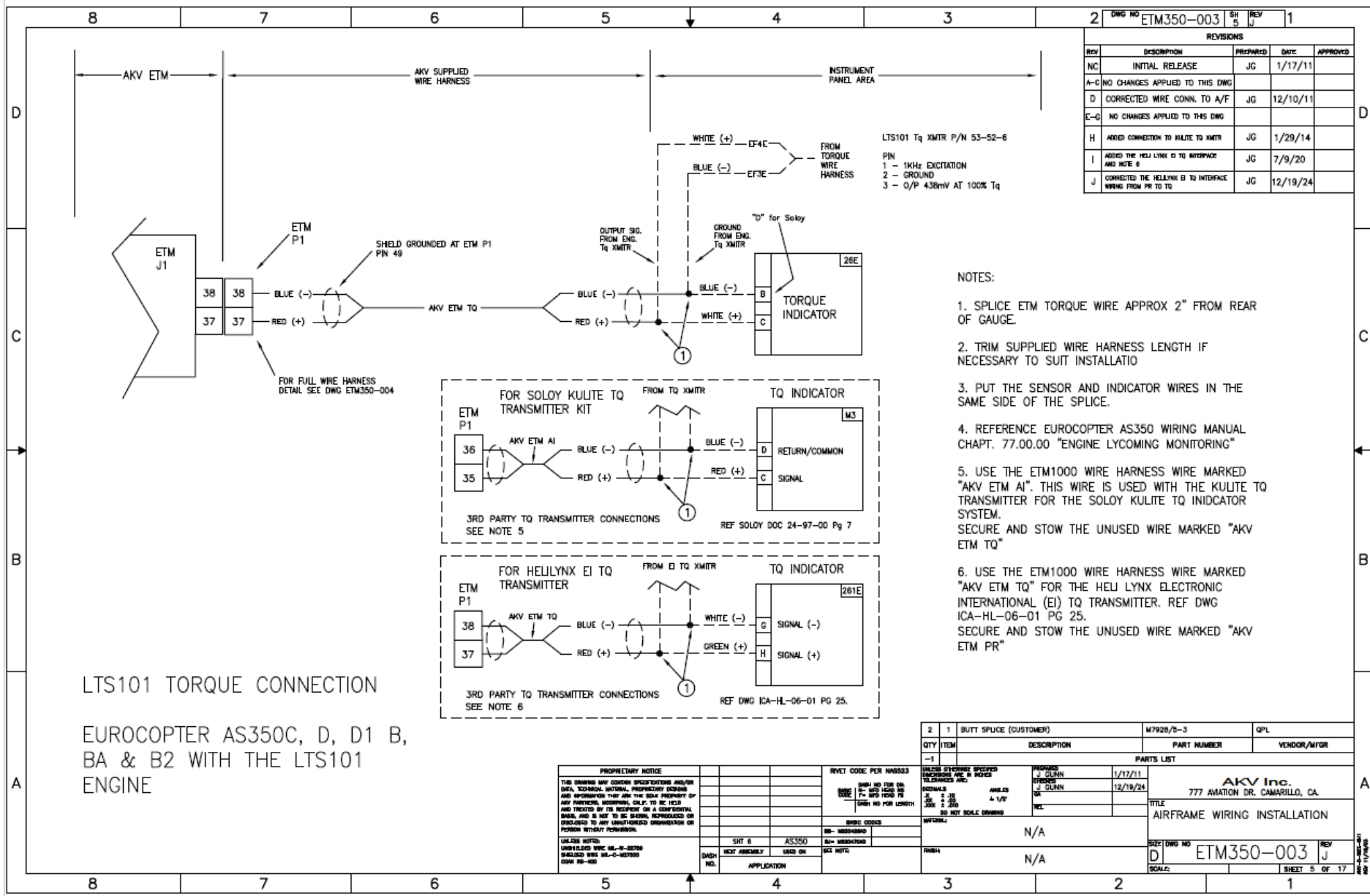
LTS101 THERMOCOUPLE CONNECTION

EUROCOPTER AS350C, D, D1 B, BA & B2 WITH THE LTS101 ENGINE

PROPRIETARY NOTICE THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGNS AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.		RIVET CODE PER NAS523 DASH NO FOR DIA 8000 8= MFD HEAD HS 1000 1= MFD HEAD FS DASH NO FOR LENGTH 1= 1/8" 2= 1/4" 3= 3/8" 4= 1/2" 5= 5/8" 6= 3/4" 7= 7/8" 8= 1"	
UNLESS NOTED: UNSHIELDED WIRE MIL-W-22758 SHIELDED WIRE MIL-C-467800 COAX RD-400		BASIC CODES 80= MS20454D 81= MS20470D SEE NOTE:	
DASH NO.	APPLICATION	SHIT 4	AS350
		NEXT ASSEMBLY	USED ON

1	1	THERMOCOUPLE SPLICE	A132439-ND	AKV/DIGKEY
QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFR
-1				
PARTS LIST				
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE:		PREPARED	1/17/11	
DECIMALS ± .010		J GUNN		
ANGLES ± 1/2°		J GUNN	7/1/20	
TAP ± .010				
DO NOT SCALE DRAWING				
MATERIAL:		N/A		
FINISH:		N/A		
TITLE		AKV Inc. 777 AVIATION DR. CAMARILLO, CA.		
SIZE		D		
DWG NO		ETM350-003		
SCALE:		SHEET 3 OF 16		

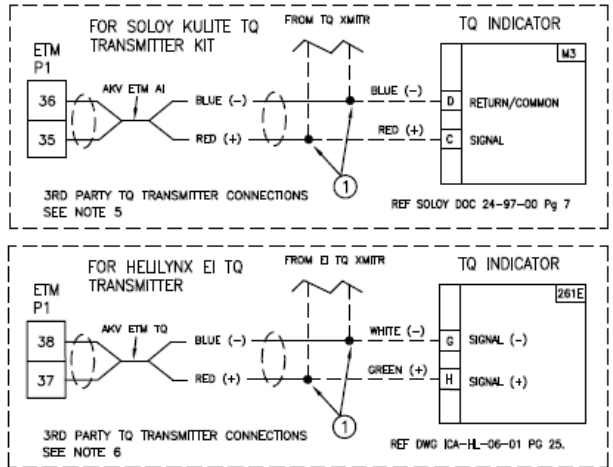




REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A-C	NO CHANGES APPLIED TO THIS DWG			
D	CORRECTED WIRE CONN. TO A/F	JG	12/10/11	
E-G	NO CHANGES APPLIED TO THIS DWG			
H	ADDED CONNECTION TO KULITE TO XMTR	JG	1/29/14	
I	ADDED THE HELI LINK ID TO INTERFACE AND NOTE 6	JG	7/9/20	
J	CORRECTED THE HELI LINK ID TO INTERFACE WIRING FROM PR TO TQ	JG	12/19/24	

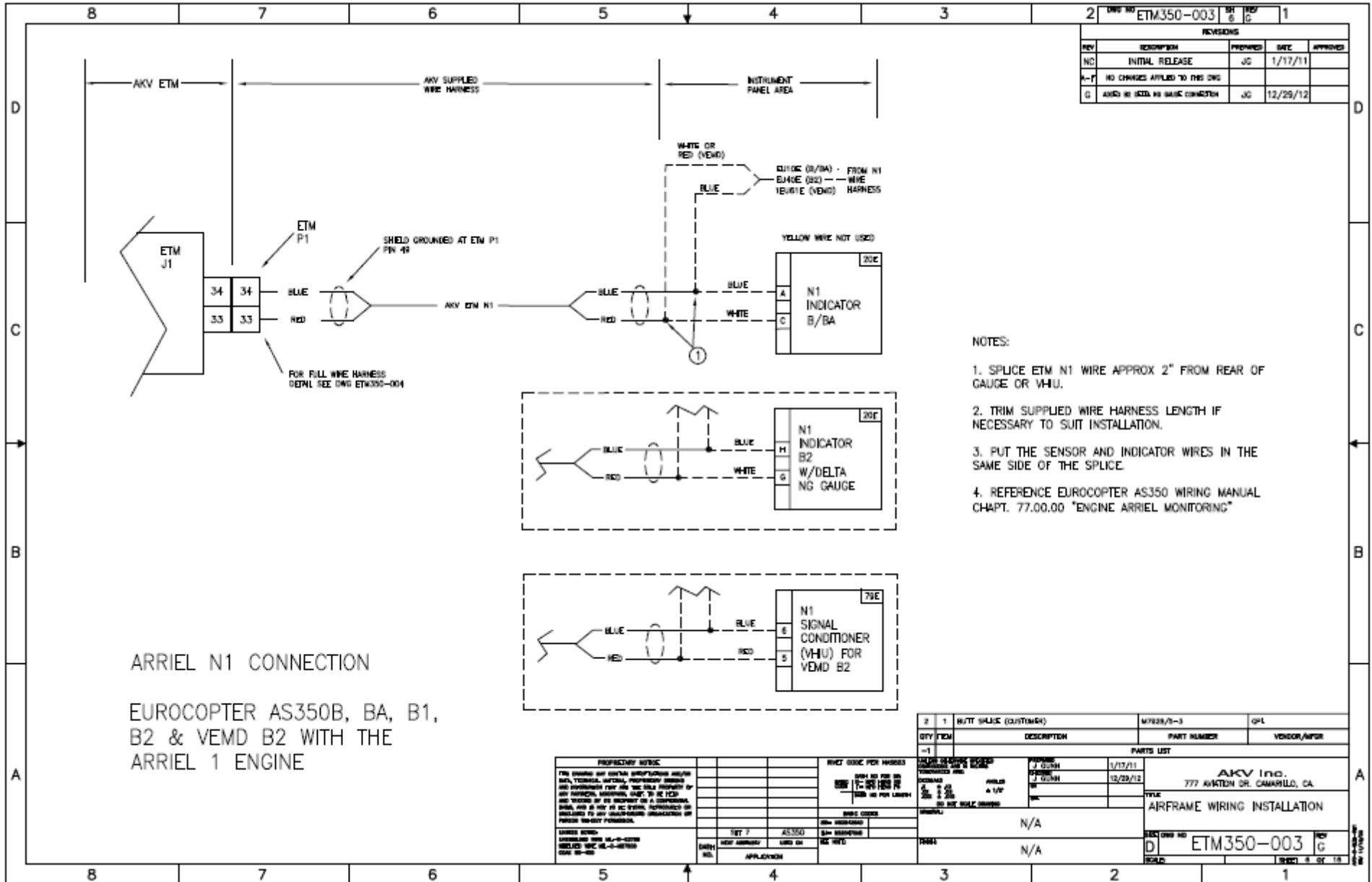
- NOTES:
1. SPLICE ETM TORQUE WIRE APPROX 2" FROM REAR OF GAUGE.
 2. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATIO
 3. PUT THE SENSOR AND INDICATOR WIRES IN THE SAME SIDE OF THE SPICE.
 4. REFERENCE EUROCOPTER AS350 WIRING MANUAL CHAPT. 77.00.00 "ENGINE LYCOMING MONITORING"
 5. USE THE ETM1000 WIRE HARNESS WIRE MARKED "AKV ETM AI". THIS WIRE IS USED WITH THE KULITE TO TRANSMITTER FOR THE SOLOY KULITE TQ INDICATOR SYSTEM. SECURE AND STOW THE UNUSED WIRE MARKED "AKV ETM TQ"
 6. USE THE ETM1000 WIRE HARNESS WIRE MARKED "AKV ETM TQ" FOR THE HELI LYNX ELECTRONIC INTERNATIONAL (EI) TQ TRANSMITTER. REF DWG ICA-HL-06-01 PG 25. SECURE AND STOW THE UNUSED WIRE MARKED "AKV ETM PR"

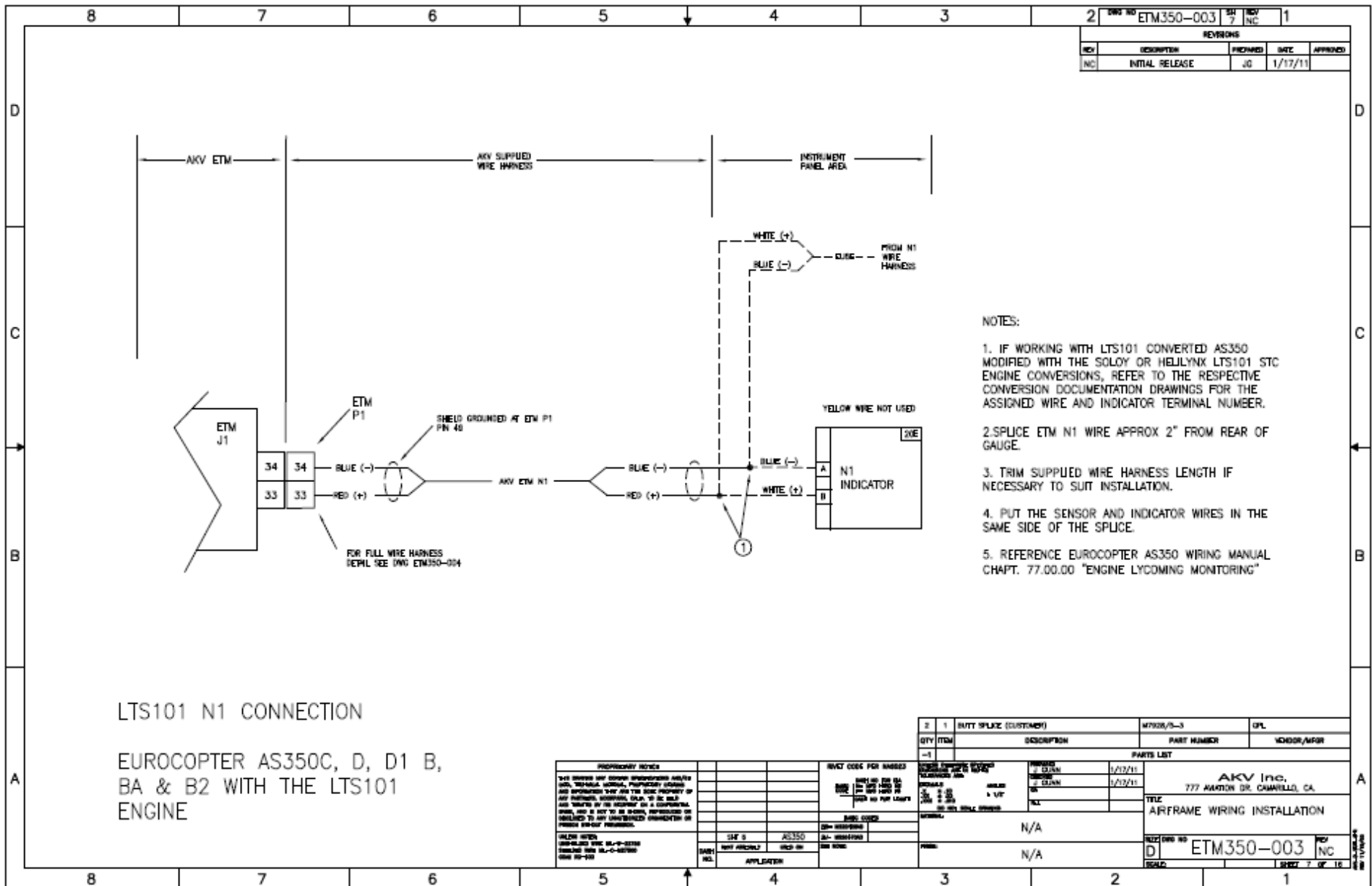
LTS101 TORQUE CONNECTION
 EUROCOPTER AS350C, D, D1 B,
 BA & B2 WITH THE LTS101
 ENGINE

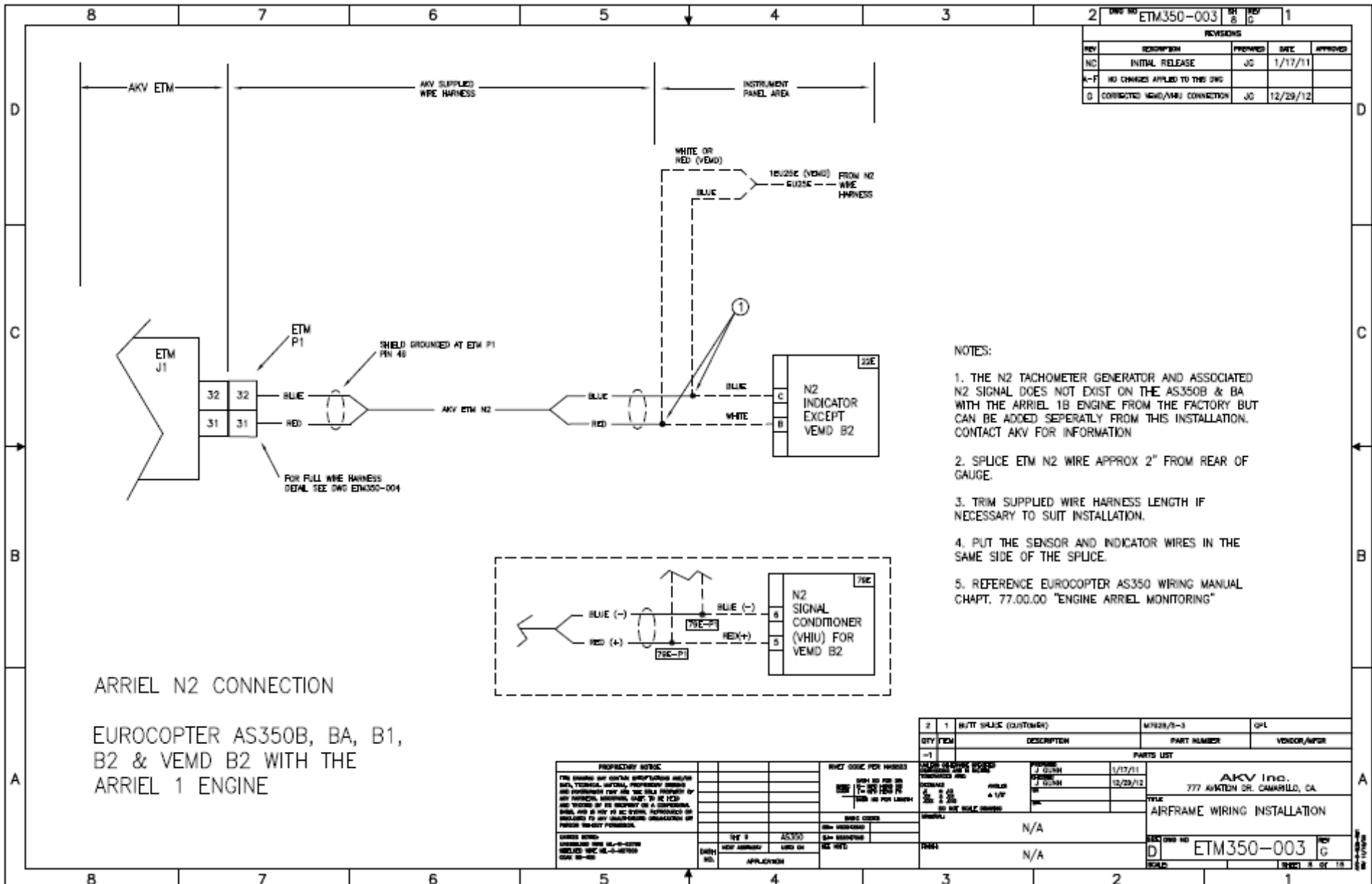


2	1	BUTT SPlice (CUSTOMER)	M7928/5-3	QPL
QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFR
-1				
PARTS LIST		PROPOSED	1/17/11	
		J GUNN		
		REVISED	12/19/24	
		J GUNN		
		OR		
		OR		
		OR		
		DO NOT SCALE DRAWING		
INFORMA:				
		N/A		
		N/A		
TITLE		AIRFRAME WIRING INSTALLATION		
DRAWN BY		AKV Inc.		
CHECKED BY		777 ANATON DR. CAMARILLO, CA.		
SCALE		D		
SHEET NO		ETM350-003		
REV		J		
SHEET 5 OF 17				

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DASH NO. SHEET NO. APPLICATION	AS350 USED ON HELI ASSEMBLY	SHW NO 538 86 SHW NO 539 86 SHW NO 540 86 SHW NO 541 86 SHW NO 542 86 SHW NO 543 86 SHW NO 544 86 SHW NO 545 86 SHW NO 546 86 SHW NO 547 86 SHW NO 548 86 SHW NO 549 86 SHW NO 550 86 SHW NO 551 86 SHW NO 552 86 SHW NO 553 86 SHW NO 554 86 SHW NO 555 86 SHW NO 556 86 SHW NO 557 86 SHW NO 558 86 SHW NO 559 86 SHW NO 560 86 SHW NO 561 86 SHW NO 562 86 SHW NO 563 86 SHW NO 564 86 SHW NO 565 86 SHW NO 566 86 SHW NO 567 86 SHW NO 568 86 SHW NO 569 86 SHW NO 570 86 SHW NO 571 86 SHW NO 572 86 SHW NO 573 86 SHW NO 574 86 SHW NO 575 86 SHW NO 576 86 SHW NO 577 86 SHW NO 578 86 SHW NO 579 86 SHW NO 580 86 SHW NO 581 86 SHW NO 582 86 SHW NO 583 86 SHW NO 584 86 SHW NO 585 86 SHW NO 586 86 SHW NO 587 86 SHW NO 588 86 SHW NO 589 86 SHW NO 590 86 SHW NO 591 86 SHW NO 592 86 SHW NO 593 86 SHW NO 594 86 SHW NO 595 86 SHW NO 596 86 SHW NO 597 86 SHW NO 598 86 SHW NO 599 86 SHW NO 600 86







REVISIONS				
REV	DESCRIPTION	APPROVED	DATE	APPROVED
ND	INITIAL RELEASE	JG	1/17/11	
K-F	NO CHANGES APPLIED TO THIS DWG			
D	CORRECTED WEMD/VEMD CONNECTION	JG	12/29/12	

- NOTES:
1. THE N2 TACHOMETER GENERATOR AND ASSOCIATED N2 SIGNAL DOES NOT EXIST ON THE AS350B & BA WITH THE ARRIEL 1B ENGINE FROM THE FACTORY BUT CAN BE ADDED SEPARATELY FROM THIS INSTALLATION. CONTACT AKV FOR INFORMATION
 2. SPLICE ETM N2 WIRE APPROX 2" FROM REAR OF GAUGE.
 3. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATION.
 4. PUT THE SENSOR AND INDICATOR WIRES IN THE SAME SIDE OF THE SPLICE.
 5. REFERENCE EUROCOPTER AS350 WIRING MANUAL CHAPT. 77.00.00 "ENGINE ARRIEL MONITORING"

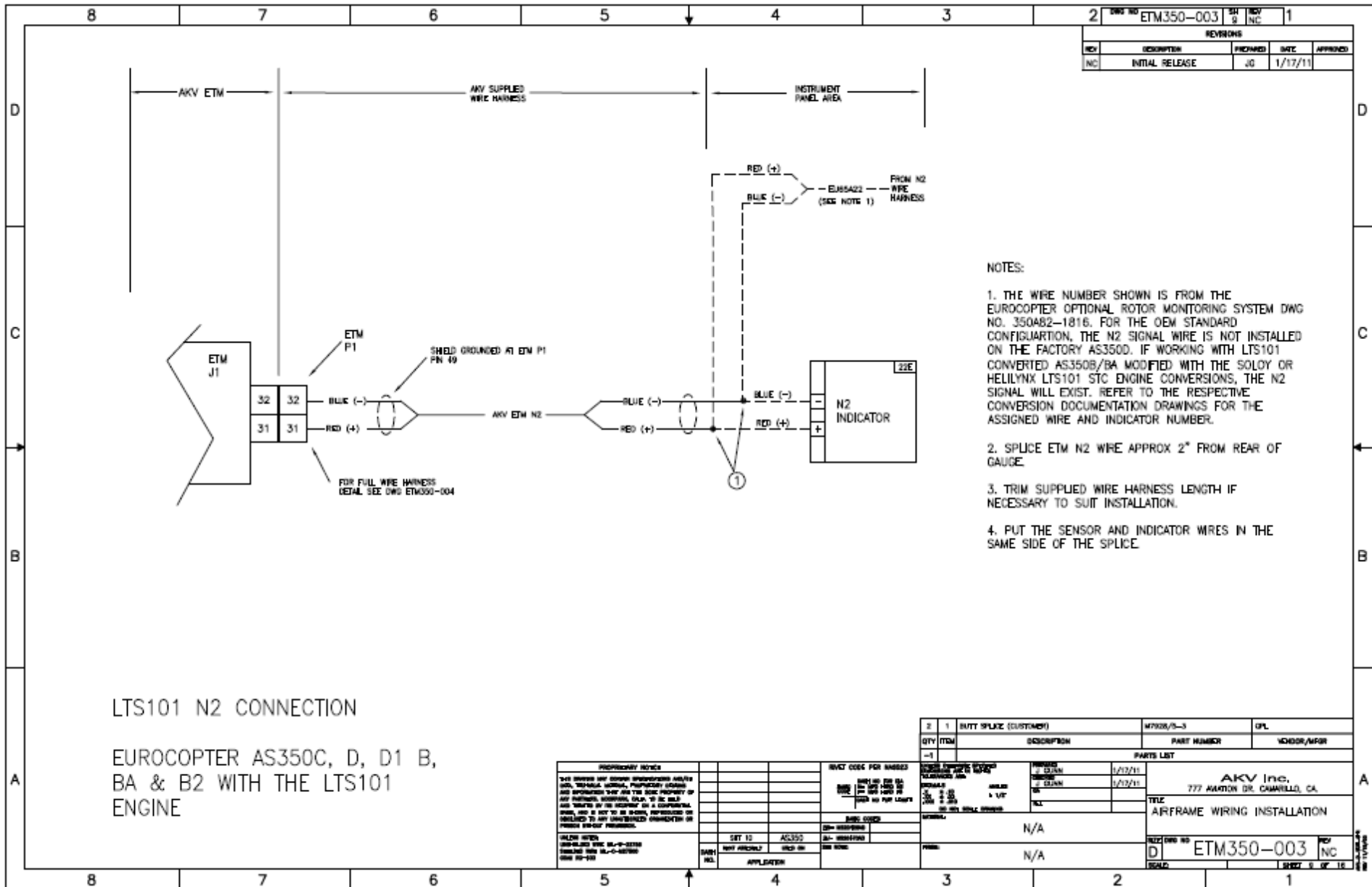
ARRIEL N2 CONNECTION
 EUROCOPTER AS350B, BA, B1,
 B2 & VEMD B2 WITH THE
 ARRIEL 1 ENGINE

PROBATIONARY NOTICE		RWC CODE FOR AIRWORTHINESS	
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UNLESS OTHERWISE SPECIFIED, ALL DIMENSIONS ARE IN INCHES AND DECIMALS THEREOF.		SEE INSTRUCTIONS FOR WIRING	
ALL DIMENSIONS ARE TO UNLESS OTHERWISE SPECIFIED.		SEE INSTRUCTIONS FOR WIRING	
ALL DIMENSIONS ARE TO UNLESS OTHERWISE SPECIFIED.		SEE INSTRUCTIONS FOR WIRING	

QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MPN
1		NUTT SLEEVE (CUSTOMER)	M7928/3-3	QPL

PARTS LIST	
1	N/A
2	N/A
3	N/A
4	N/A
5	N/A

AKV Inc. 777 AVIATION DR. CAMARILLO, CA	
AIRFRAME WIRING INSTALLATION	
DWG NO: ETM350-003	REV: F
DATE: 12/29/12	BY: JG



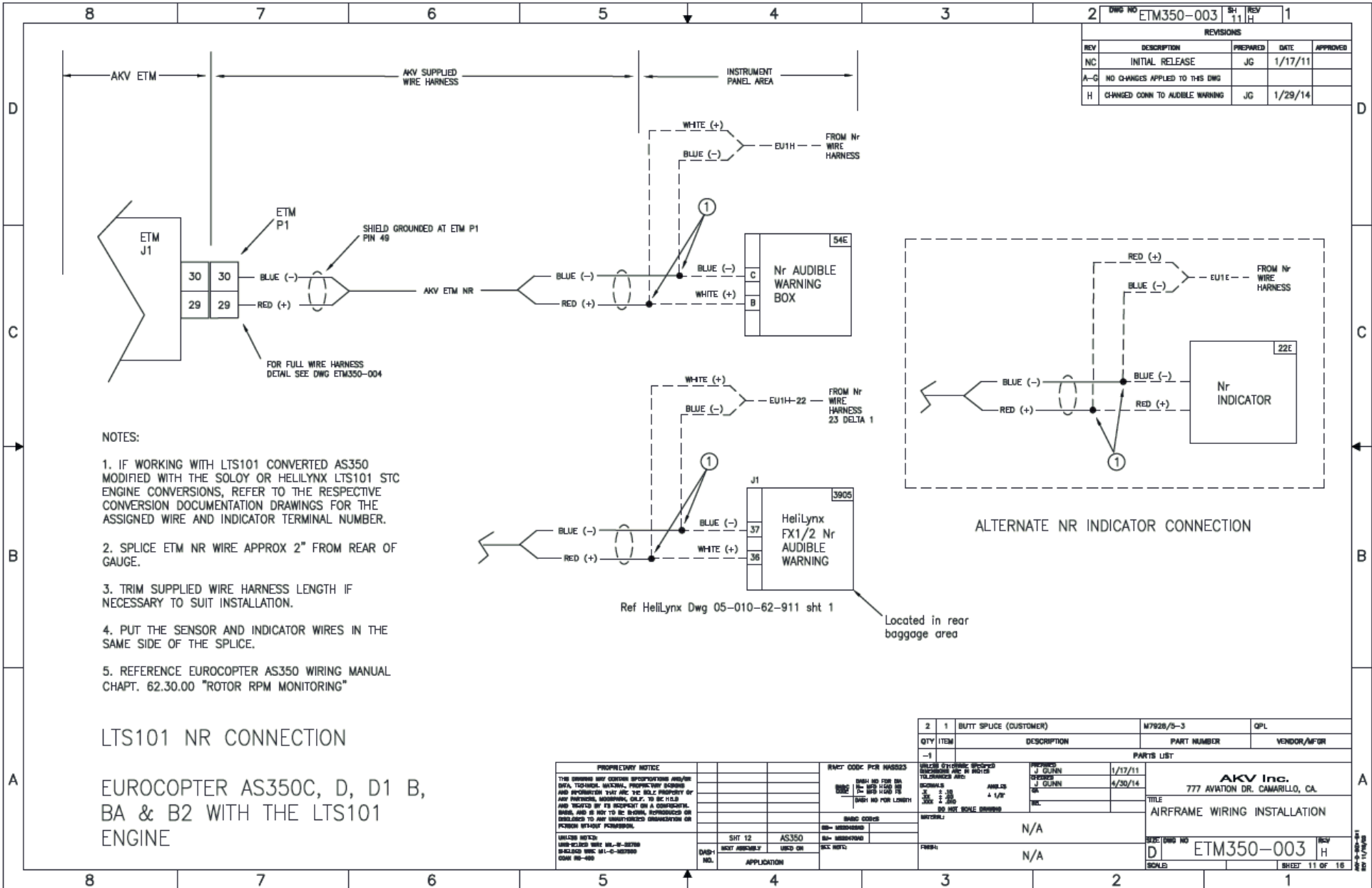
2	DWG NO	ETM350-003	REV	INC	1
REVISIONS					
REV	DESCRIPTION	PREPARED	DATE	APPROVED	
NC	INITIAL RELEASE	JG	1/17/11		

- NOTES:
1. THE WIRE NUMBER SHOWN IS FROM THE EUROCOPTER OPTIONAL ROTOR MONITORING SYSTEM DWG NO. 350482-1816. FOR THE OEM STANDARD CONFIGURATION, THE N2 SIGNAL WIRE IS NOT INSTALLED ON THE FACTORY AS350D. IF WORKING WITH LTS101 CONVERTED AS350B/BA MODIFIED WITH THE SOLOY OR HELILYNX LTS101 STC ENGINE CONVERSIONS, THE N2 SIGNAL WILL EXIST. REFER TO THE RESPECTIVE CONVERSION DOCUMENTATION DRAWINGS FOR THE ASSIGNED WIRE AND INDICATOR NUMBER.
 2. SPLICE ETM N2 WIRE APPROX 2" FROM REAR OF GAUGE.
 3. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATION.
 4. PUT THE SENSOR AND INDICATOR WIRES IN THE SAME SIDE OF THE SPLICE.

LTS101 N2 CONNECTION
 EUROCOPTER AS350C, D, D1 B,
 BA & B2 WITH THE LTS101
 ENGINE

QTY		1	1	1	1
ITEM		BUTT SPLICE (CUSTOMER)		W7928/S-3	QPL
DESCRIPTION				PART NUMBER	1000R/MFR
PARTS LIST					
REV		1		1/2/21	
DATE		1/2/21			
BY		AKV			
CHK		AKV			
APP		AKV			
TITLE		AKV Inc. 777 AVATION DR. CAMARILLO, CA			
TITLE		AIRFRAME WIRING INSTALLATION			
REV		D		1/2/21	
DWG NO		ETM350-003		REV	INC
SCALE					
SHEET		9		OF	16

REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A-C	NO CHANGES APPLIED TO THIS DWG			
H	CHANGED CONN TO AUDIBLE WARNING	JG	1/29/14	



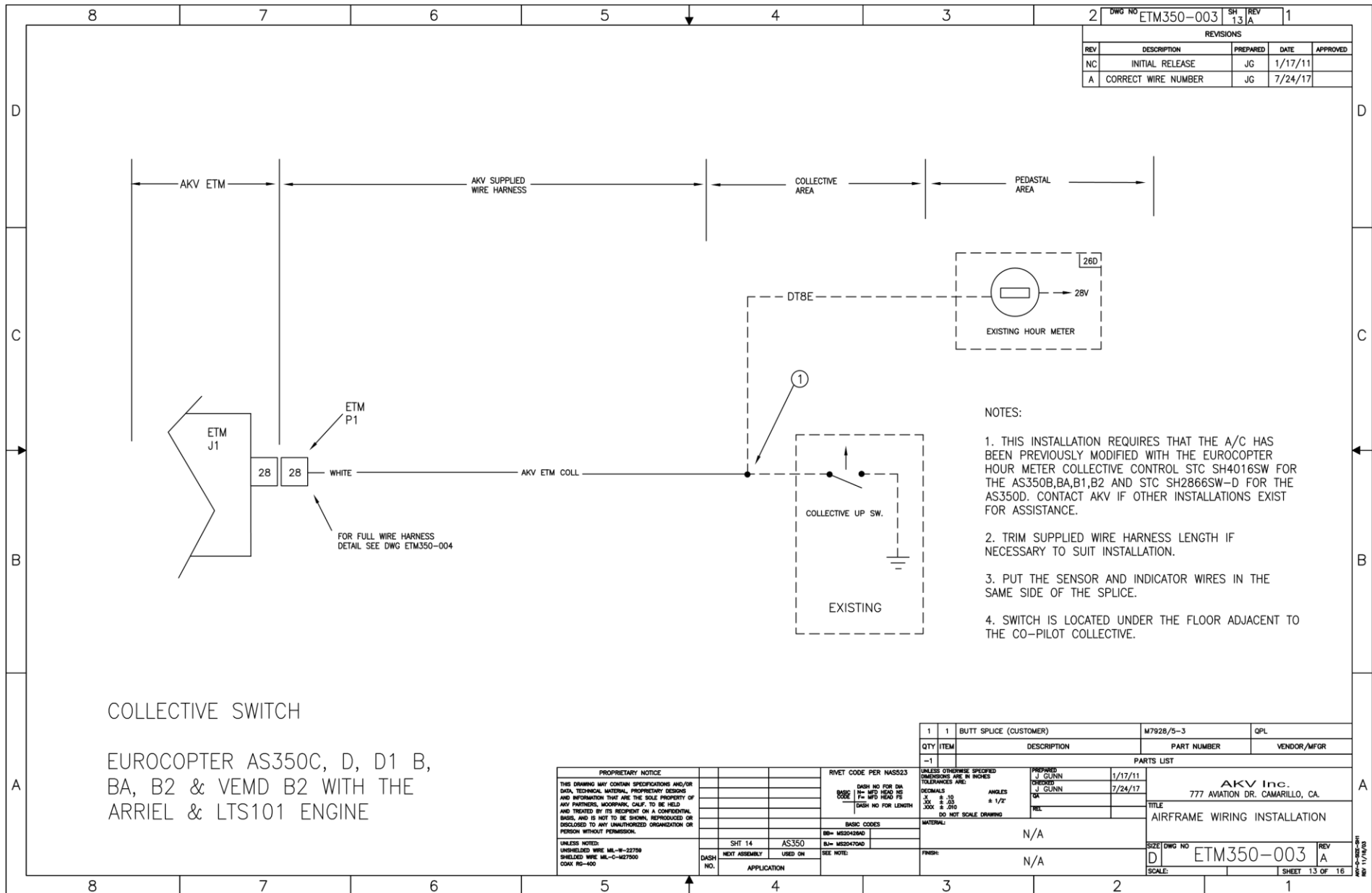
NOTES:

1. IF WORKING WITH LTS101 CONVERTED AS350 MODIFIED WITH THE SOLOY OR HELLYNX LTS101 STC ENGINE CONVERSIONS, REFER TO THE RESPECTIVE CONVERSION DOCUMENTATION DRAWINGS FOR THE ASSIGNED WIRE AND INDICATOR TERMINAL NUMBER.
2. SPLICE ETM NR WIRE APPROX 2" FROM REAR OF GAUGE.
3. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATION.
4. PUT THE SENSOR AND INDICATOR WIRES IN THE SAME SIDE OF THE SPLICE.
5. REFERENCE EUROCOPTER AS350 WIRING MANUAL CHAPT. 62.30.00 "ROTOR RPM MONITORING"

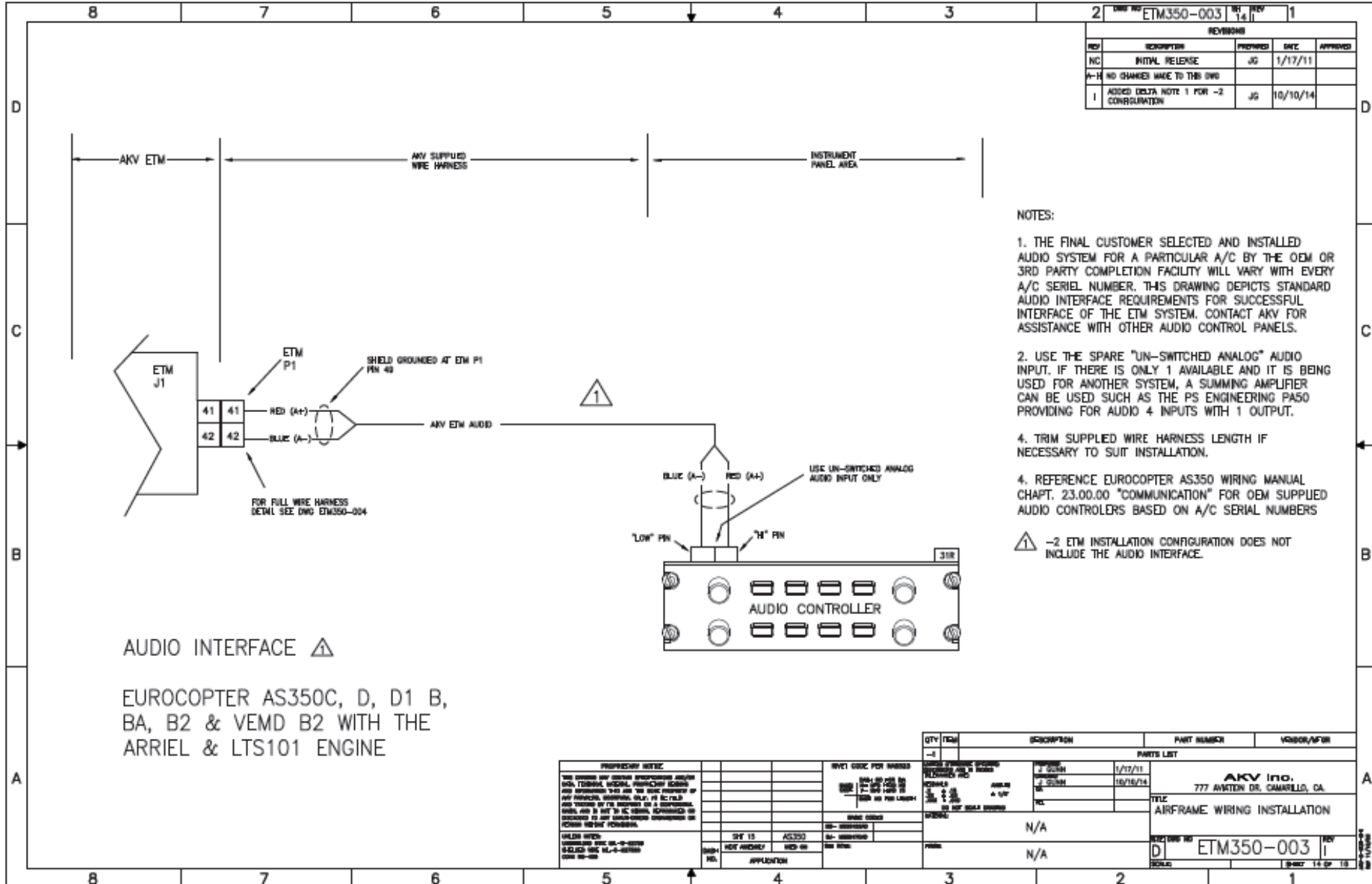
LTS101 NR CONNECTION

EUROCOPTER AS350C, D, D1 B, BA & B2 WITH THE LTS101 ENGINE

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UNLESS NOTED OTHERWISE, DIMENSIONS ARE IN INCHES. DIMENSIONS IN PARENTHESES ARE IN MILLIMETERS. DIMENSIONS IN BRACKETS ARE FOR REFERENCE ONLY. DIMENSIONS IN SQUARE BRACKETS ARE FOR INFORMATION ONLY. DIMENSIONS IN CIRCLES ARE FOR INFORMATION ONLY. DIMENSIONS IN TRIANGLES ARE FOR INFORMATION ONLY. DIMENSIONS IN DIAMOND SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN PARALLELOGRAMS ARE FOR INFORMATION ONLY. DIMENSIONS IN TRAPEZOIDAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN OVAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN STAR SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN HEXAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN OCTAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN CIRCULAR SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN TRIANGULAR SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN QUADRILATERAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN PENTAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN HEXAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN SEPTAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN OCTAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN NONAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN DECAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN UNDECAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN DODECAGONAL SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN TRICORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN QUADRICORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN PENTACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN HEXACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN SEPTACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN OCTACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN ENNEACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN DECACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN UNDECACORNER SHAPES ARE FOR INFORMATION ONLY. DIMENSIONS IN DODECACORNER SHAPES ARE FOR INFORMATION ONLY.	SHEET NO: 11 OF 16 DWG NO: ETM350-003 REV: H			



REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A-H	NO CHANGES MADE TO THIS DWG			
1	ADDED DELTA NOTE 1 FOR -2 CONFIGURATION	JG	10/10/14	

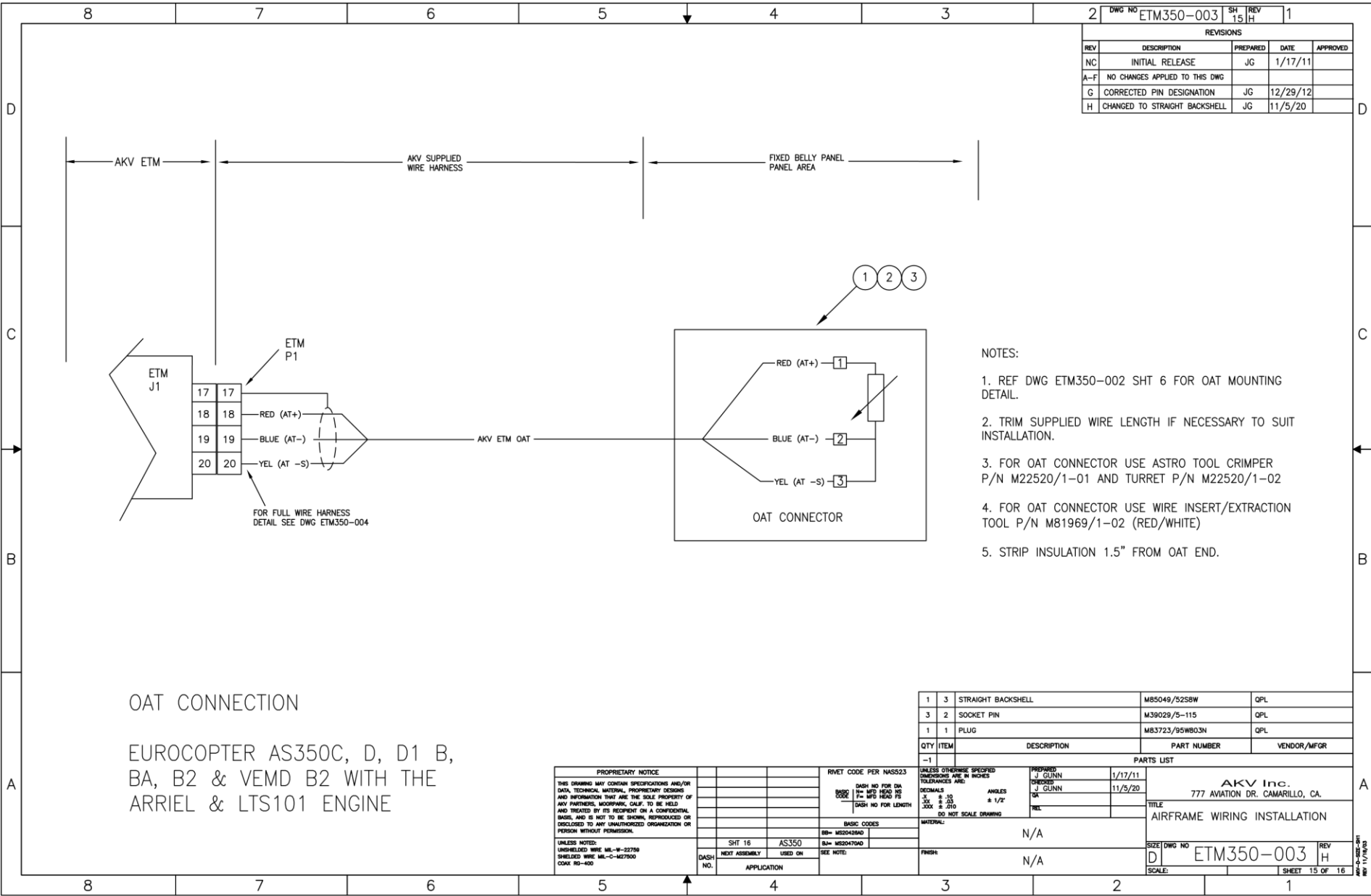


- NOTES:
1. THE FINAL CUSTOMER SELECTED AND INSTALLED AUDIO SYSTEM FOR A PARTICULAR A/C BY THE OEM OR 3RD PARTY COMPLETION FACILITY WILL VARY WITH EVERY A/C SERIAL NUMBER. THIS DRAWING DEPICTS STANDARD AUDIO INTERFACE REQUIREMENTS FOR SUCCESSFUL INTERFACE OF THE ETM SYSTEM. CONTACT AKV FOR ASSISTANCE WITH OTHER AUDIO CONTROL PANELS.
 2. USE THE SPARE "UN-SWITCHED ANALOG" AUDIO INPUT. IF THERE IS ONLY 1 AVAILABLE AND IT IS BEING USED FOR ANOTHER SYSTEM, A SUMMING AMPLIFIER CAN BE USED SUCH AS THE PS ENGINEERING PASO PROVIDING FOR AUDIO 4 INPUTS WITH 1 OUTPUT.
 4. TRIM SUPPLIED WIRE HARNESS LENGTH IF NECESSARY TO SUIT INSTALLATION.
 4. REFERENCE EUROCOPTER AS350 WIRING MANUAL CHAPT. 23.00.00 "COMMUNICATION" FOR OEM SUPPLIED AUDIO CONTROLLERS BASED ON A/C SERIAL NUMBERS
- ⚠️ -2 ETM INSTALLATION CONFIGURATION DOES NOT INCLUDE THE AUDIO INTERFACE.

AUDIO INTERFACE ⚠️
 EUROCOPTER AS350C, D, D1 B,
 BA, B2 & VEMD B2 WITH THE
 ARRIEL & LTS101 ENGINE

QTY	ITEM	DESCRIPTION	PART NUMBER	WIRING/AFIR
1				
PARTS LIST				
PREPARED		J. GUNN	1/17/11	
CHECKED		J. GUNN	10/10/14	
APPROVED				
DATE				
BY				
FOR				
TITLE		AIRFRAME WIRING INSTALLATION		
DWG NO		ETM350-003		
REV		1		
SCALE		1:1		

REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	1/17/11	
A-F	NO CHANGES APPLIED TO THIS DWG			
G	CORRECTED PIN DESIGNATION	JG	12/29/12	
H	CHANGED TO STRAIGHT BACKSHELL	JG	11/5/20	



NOTES:

1. REF DWG ETM350-002 SHT 6 FOR OAT MOUNTING DETAIL.
2. TRIM SUPPLIED WIRE LENGTH IF NECESSARY TO SUIT INSTALLATION.
3. FOR OAT CONNECTOR USE ASTRO TOOL CRIMPER P/N M22520/1-01 AND TURRET P/N M22520/1-02
4. FOR OAT CONNECTOR USE WIRE INSERT/EXTRACTION TOOL P/N M81969/1-02 (RED/WHITE)
5. STRIP INSULATION 1.5" FROM OAT END.

OAT CONNECTION

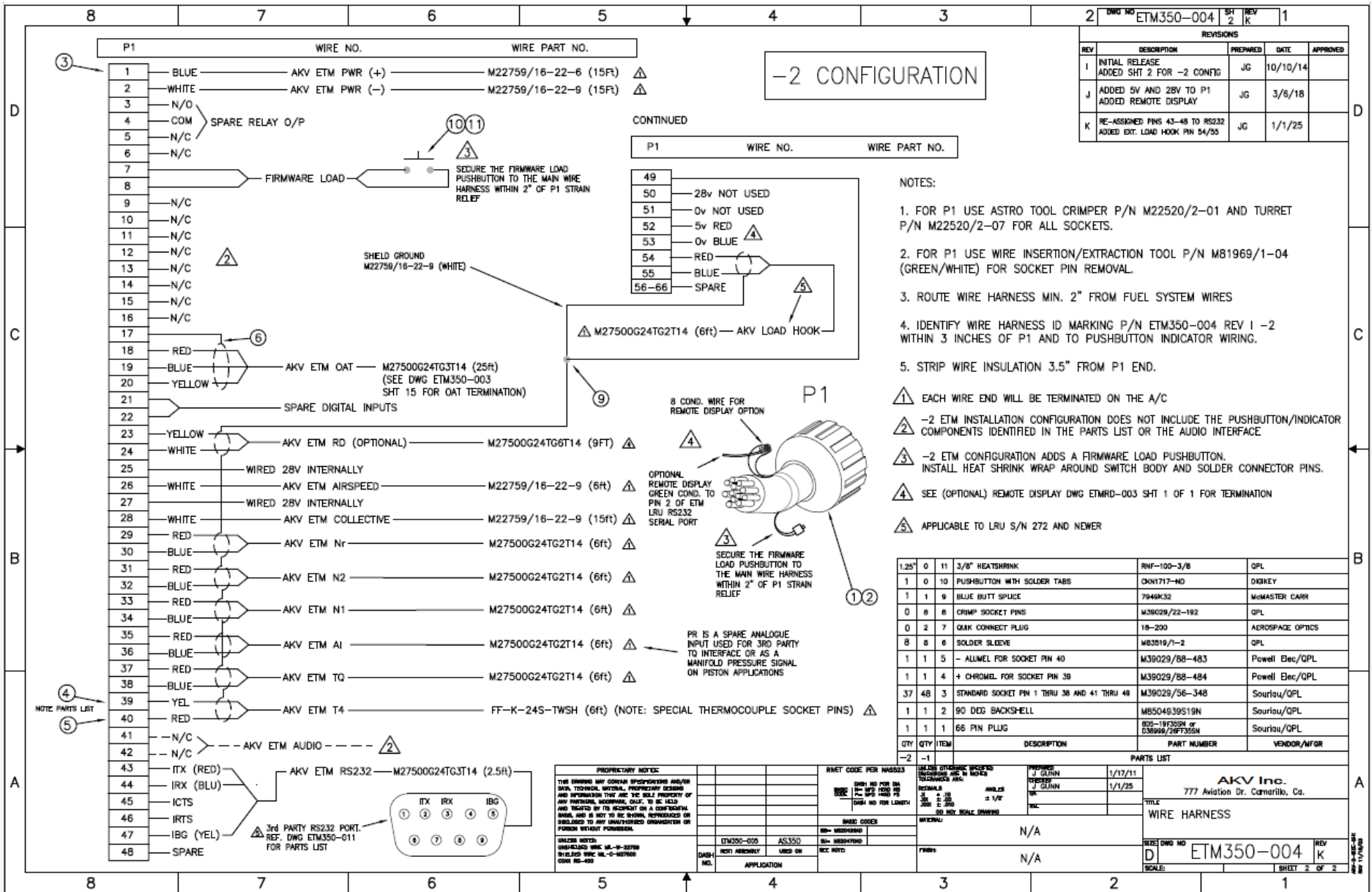
EUROCOPTER AS350C, D, D1 B,
 BA, B2 & VEMD B2 WITH THE
 ARIEL & LTS101 ENGINE

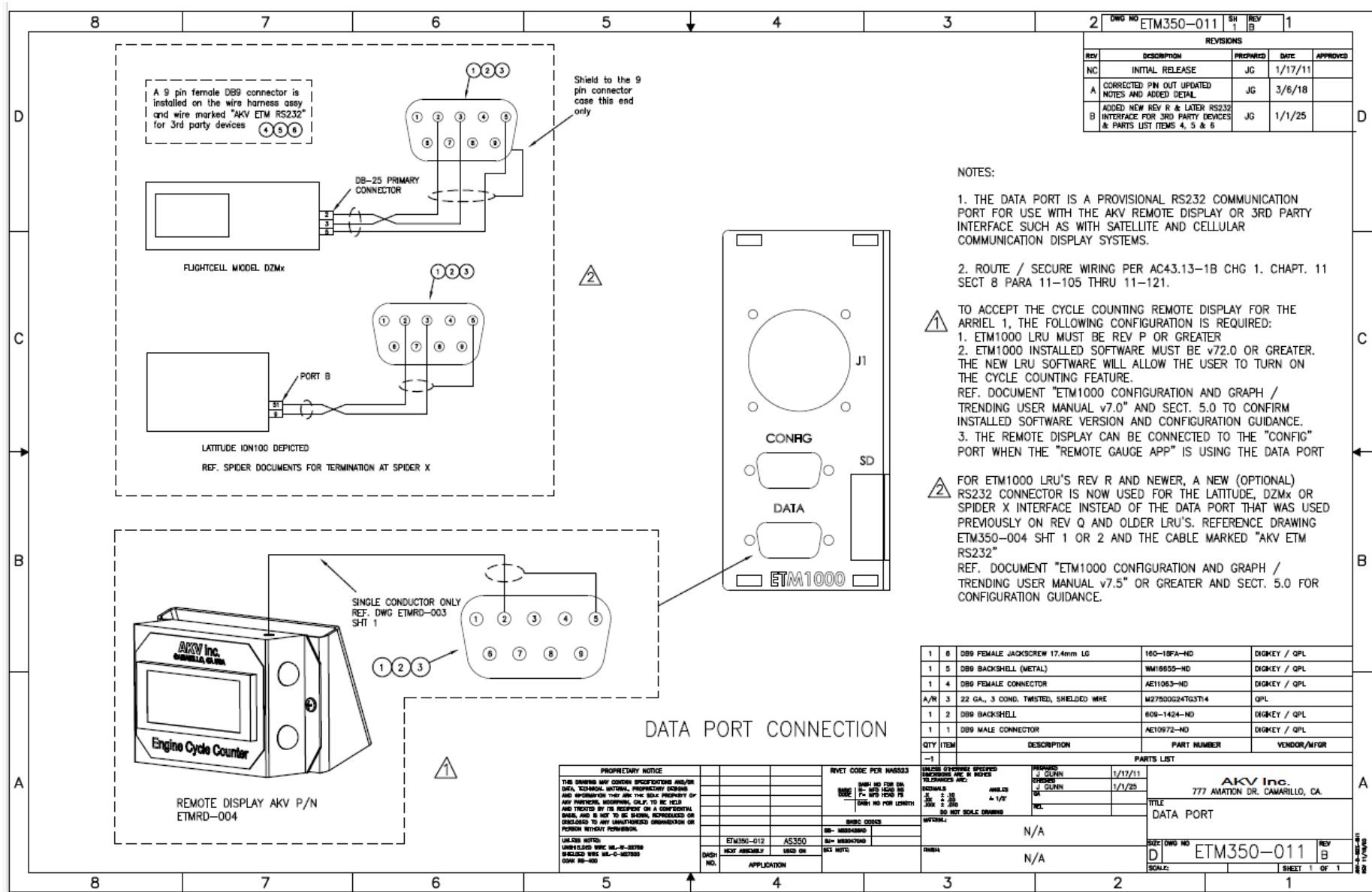
PROPRIETARY NOTICE <small>THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGN AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.</small>		<small>UNLESS NOTED: UNSHIELDED WIRE MIL-W-22758 SHIELDED WIRE MIL-C-127700 CDAX RG-400</small>	
<small>THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGN AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.</small>	<small>UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE: DECIMALS .010 FRACTIONS 1/32 ANGLES ± 1/2° HOLE POSITION ± .010 DO NOT SCALE DRAWING</small>	<small>PREPARED 1/17/11 CHECKED 11/5/20</small>	<small>AKV Inc. 777 AVIATION DR. CAMARILLO, CA.</small>
<small>UNLESS NOTED: UNSHIELDED WIRE MIL-W-22758 SHIELDED WIRE MIL-C-127700 CDAX RG-400</small>	<small>THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGN AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.</small>	<small>UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE: DECIMALS .010 FRACTIONS 1/32 ANGLES ± 1/2° HOLE POSITION ± .010 DO NOT SCALE DRAWING</small>	<small>AKV Inc. 777 AVIATION DR. CAMARILLO, CA.</small>
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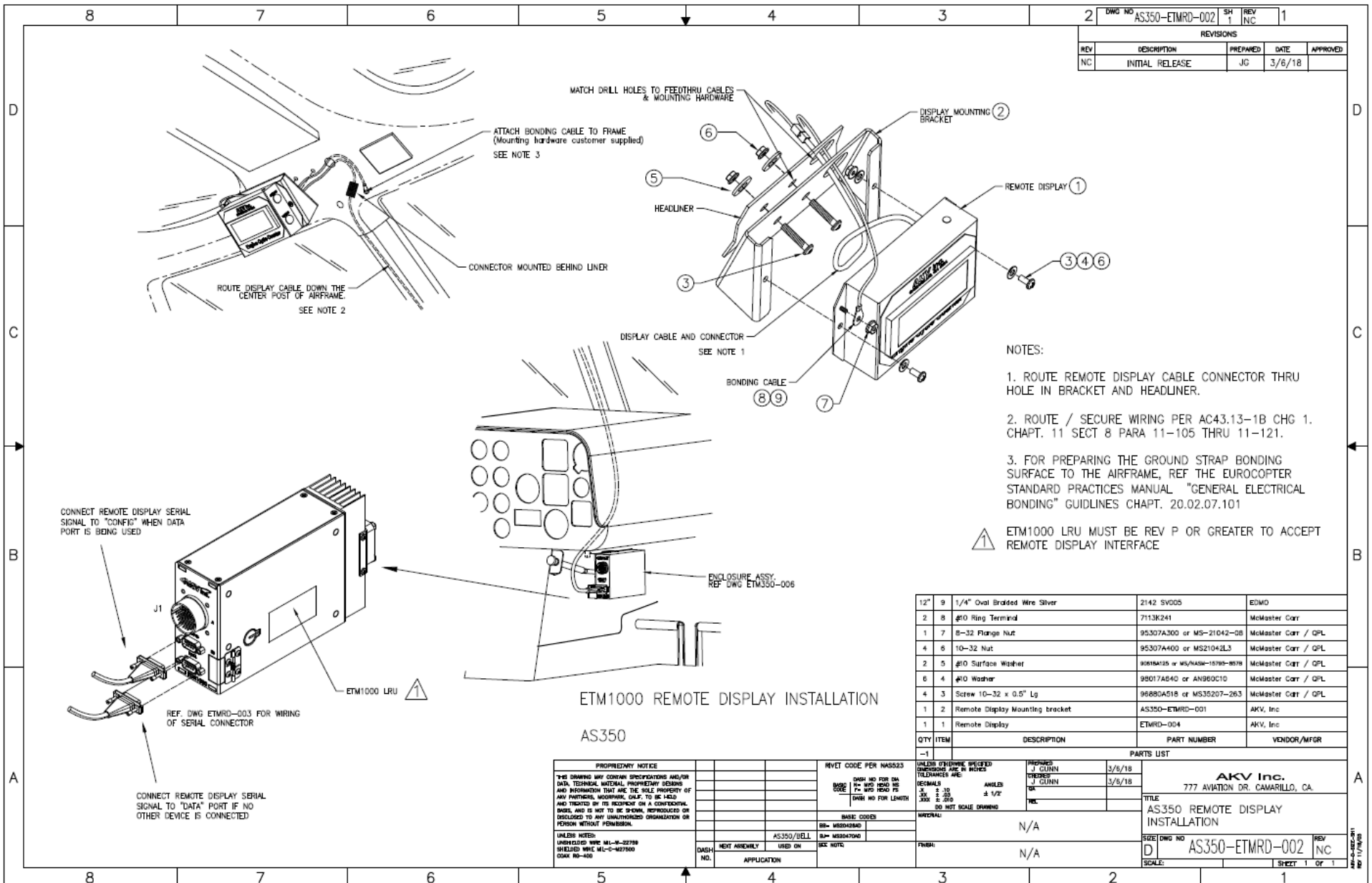
QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFOR
1	3	STRAIGHT BACKSHELL	M85049/52SBW	QPL
3	2	SOCKET PIN	M39029/5-115	QPL
1	1	PLUG	M83723/95WB03N	QPL

PARTS LIST		DATE	TITLE
PREPARED	1/17/11		
CHECKED	11/5/20		
MATERIAL:			
FINISH:			
N/A			
N/A			

<small>UNLESS NOTED: UNSHIELDED WIRE MIL-W-22758 SHIELDED WIRE MIL-C-127700 CDAX RG-400</small>	<small>THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGN AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.</small>	<small>UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE: DECIMALS .010 FRACTIONS 1/32 ANGLES ± 1/2° HOLE POSITION ± .010 DO NOT SCALE DRAWING</small>	<small>AKV Inc. 777 AVIATION DR. CAMARILLO, CA.</small>
<small>UNLESS NOTED: UNSHIELDED WIRE MIL-W-22758 SHIELDED WIRE MIL-C-127700 CDAX RG-400</small>	<small>THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGN AND INFORMATION THAT ARE THE SOLE PROPERTY OF AKV PARTNERS, MOORPARK, CALIF. TO BE HELD AND TREATED BY ITS RECIPIENT ON A CONFIDENTIAL BASIS, AND IS NOT TO BE SHOWN, REPRODUCED OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.</small>	<small>UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ARE: DECIMALS .010 FRACTIONS 1/32 ANGLES ± 1/2° HOLE POSITION ± .010 DO NOT SCALE DRAWING</small>	<small>AKV Inc. 777 AVIATION DR. CAMARILLO, CA.</small>







REVISIONS				
REV	DESCRIPTION	PREPARED	DATE	APPROVED
NC	INITIAL RELEASE	JG	3/6/18	

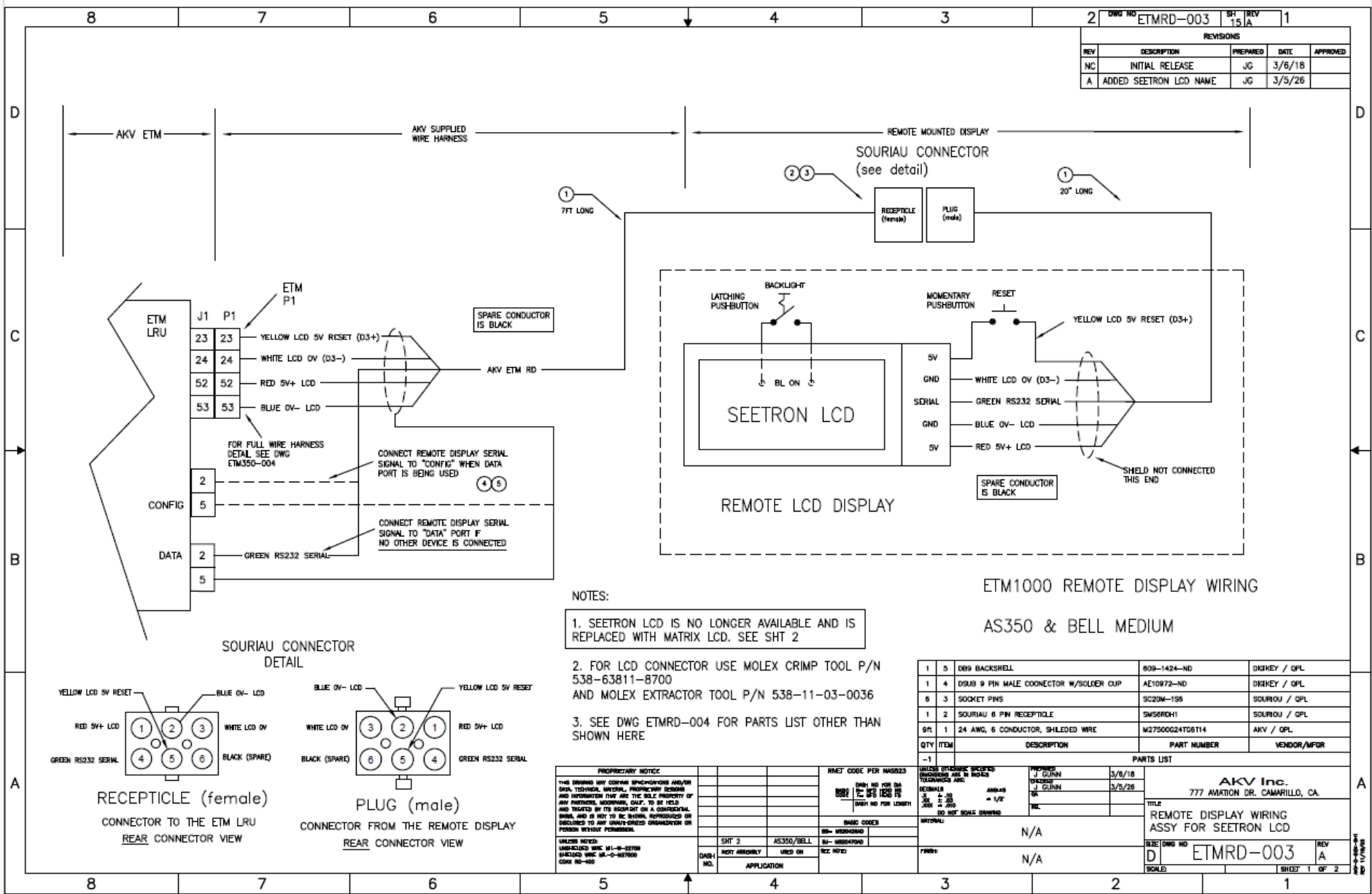
- NOTES:
1. ROUTE REMOTE DISPLAY CABLE CONNECTOR THRU HOLE IN BRACKET AND HEADLINER.
 2. ROUTE / SECURE WIRING PER AC43.13-1B CHG 1. CHAPT. 11 SECT 8 PARA 11-105 THRU 11-121.
 3. FOR PREPARING THE GROUND STRAP BONDING SURFACE TO THE AIRFRAME, REF THE EUROCOPTER STANDARD PRACTICES MANUAL "GENERAL ELECTRICAL BONDING" GUIDELINES CHAPT. 20.02.07.101

⚠ ETM1000 LRU MUST BE REV P OR GREATER TO ACCEPT REMOTE DISPLAY INTERFACE

QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MFOR
12"	9	1/4" Oval Braided Wire Silver	2142 SV005	EDMO
2	8	#10 Ring Terminal	7113K241	McMaster Carr
1	7	8-32 Flange Nut	95307A300 or MS-21042-08	McMaster Carr / OPL
4	6	10-32 Nut	95307A400 or MS21042L3	McMaster Carr / OPL
2	5	#10 Surface Washer	9081M25 or MS/NASB-15780-8978	McMaster Carr / OPL
6	4	#10 Washer	98017A840 or AN960C10	McMaster Carr / OPL
4	3	Screw 10-32 x 0.5" Lg	96880A518 or MS35207-263	McMaster Carr / OPL
1	2	Remote Display Mounting bracket	AS350-ETMRD-001	AKV, Inc
1	1	Remote Display	ETMRD-004	AKV, Inc

PROPRIETARY NOTICE		REVET CODE PER NAS553		UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES		PREPARED		DATE	
THIS DRAWING MAY CONTAIN SPECIFICATIONS AND/OR DATA, TECHNICAL MATERIAL, PROPRIETARY DESIGN AND INFORMATION THAT ARE THE SOLE PROPERTY OF ANY PATENTED, UNPATENTED, OR TRADE SECRET AND IS NOT TO BE REPRODUCED, COPIED, OR DISCLOSED TO ANY UNAUTHORIZED ORGANIZATION OR PERSON WITHOUT PERMISSION.		DASH NO FOR ON BASE CODE		DECIMALS: 3/16 ± .010 1/8 ± .015 3/32 ± .020		J CLIN		3/5/18	
UNLESS NOTED: UNREVIEWED WIRE MIL-9-22789 (SILVER) WIRE MIL-9-427500 (GOLD) R5-400		BASE CODES		MATERIAL: N/A		FINISH: N/A		TITLE: AS350 REMOTE DISPLAY INSTALLATION	
DASH NO. APPLICATION		NEXT ASSEMBLY USED ON		REV: N/A		REV: NC		DRAWING NO: AS350-ETMRD-002	
								SCALE: SHEET 1 OF 1	

2		DWG NO	ETMRD-003	SHT	REV	15	1
REVISIONS							
REV	DESCRIPTION	PREPARED	DATE	APPROVED			
NC	INITIAL RELEASE	JG	3/6/18				
A	ADDED SESTRON LCD NAME	JG	3/5/26				



- NOTES:
- SESTRON LCD IS NO LONGER AVAILABLE AND IS REPLACED WITH MATRIX LCD. SEE SHT 2
 - FOR LCD CONNECTOR USE MOLEX CRIMP TOOL P/N 538-63811-8700 AND MOLEX EXTRACTOR TOOL P/N 538-11-03-0036
 - SEE DWG ETMRD-004 FOR PARTS LIST OTHER THAN SHOWN HERE

ETM1000 REMOTE DISPLAY WIRING
 AS350 & BELL MEDIUM

QTY	ITEM	DESCRIPTION	PART NUMBER	VENDOR/MPN
1	5	DE9 BACKSHELL	809-1424-ND	DIGKEY / QPL
1	4	DSUB 9 PIN MALE CONNECTOR W/SOLDER CUP	AE10972-ND	DIGKEY / QPL
8	3	SOCKET PINS	SC20M-196	SOURIAU / QPL
1	2	SOURIAU 6 PIN RECEPTACLE	SM56RDH1	SOURIAU / QPL
8/1	1	24 AWG, 6 CONDUCTOR, SHIELED WIRE	M27500024T06T14	AKV / QPL

APPROVED		DATE	
J. GUNN		3/6/18	
J. GUNN		3/5/26	
<p>AKV Inc. 777 AVIATION DR. CAMARILLO, CA.</p>			
<p>TITLE REMOTE DISPLAY WIRING ASSY FOR SESTRON LCD</p>			
SIZE	DWG NO	REV	
D	ETMRD-003	A	
SHEET	1		OF 2

<p>PROPRIETARY NOTICE</p> <p>THIS DRAWING AND ALL INFORMATION CONTAINED HEREIN ARE THE SOLE PROPERTY OF AKV, INC. AND ARE NOT TO BE REPRODUCED OR TRANSMITTED IN ANY FORM OR BY ANY MEANS, ELECTRONIC OR MECHANICAL, INCLUDING PHOTOCOPYING, RECORDING, OR BY ANY INFORMATION STORAGE AND RETRIEVAL SYSTEM, WITHOUT THE WRITTEN PERMISSION OF AKV, INC.</p>		<p>DATE: 3/5/26</p> <p>BY: JG</p> <p>FOR: REV 15</p>
<p>UNLESS NOTED OTHERWISE, THIS DRAWING IS TO BE USED IN ACCORDANCE WITH THE REQUIREMENTS OF MIL-STD-883C METHOD 2000, TEST METHOD 2000.1, AND MIL-STD-883C METHOD 2000.2, TEST METHOD 2000.2.1.</p>	<p>SHT 2</p> <p>AS350/BELL</p>	<p>REV 15</p>

